

# Standalone LEO Doppler Positioning using proposed broadband Communication Constellation

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# LEO Position and velocity estimation Using Doppler Measurements



- **Advantages:**

- Better Anti-jamming performance due to:
  - higher received power level,
  - larger number of satellites,
  - and rapid time-varying satellite geometry.
- Better multipath decorrelation time (Minimum over all types of orbits).
- No requirement of on-board atomic clock.

- **Challenges:**

- Requires **minimum 8# of simultaneous satellites** for point positioning.
- Contradictory requirements of communication and navigation systems.
- Unavailability of Pseudo-range observables.

# System Requirements

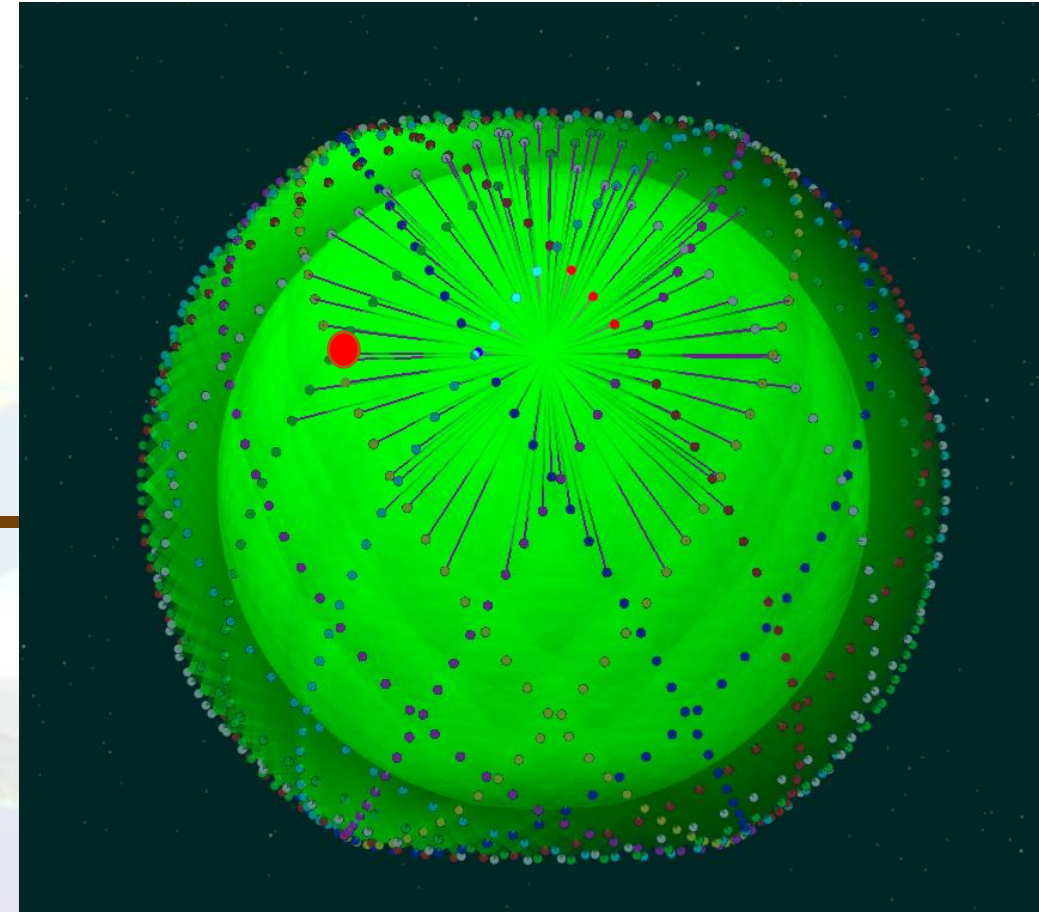


- Indian mainland coverage.
  - Targeted Position accuracy:
    - 3D RMS: < 20m
  - Velocity accuracy: <0.1 m/s
  - Range-rate measurement accuracy: < 0.01 m/s
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- Minimum number of satellite available: 8
  - Orbit determination using ground station network.
  - Orbit determination accuracy (per axis): < 3m RMS
  - Orbit velocity determination accuracy (per axis): < 0.001 m/s

# Doppler Only Position/Velocity Algorithm Development

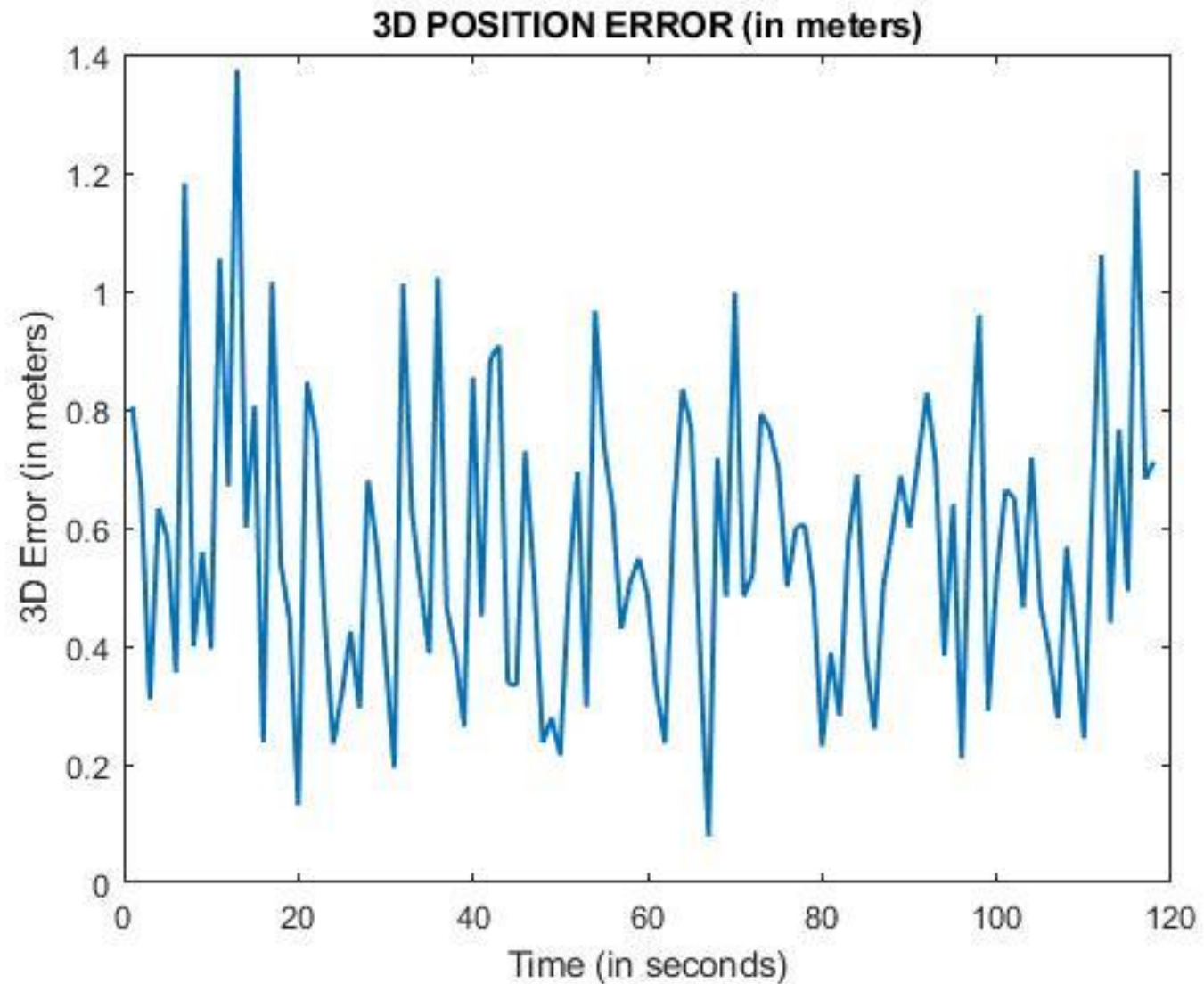


- Doppler only Positioning for Proposed LEO satellite constellation was attempted using simulated data.
- Equivalent range rate was corrupted with mean 0.1 m/sec and standard deviation of 0.003 m/sec & 0.03 m/sec.
- 3-D, Horizontal & Vertical position accuracies were obtained using all 59 visible satellite to a stationary user.



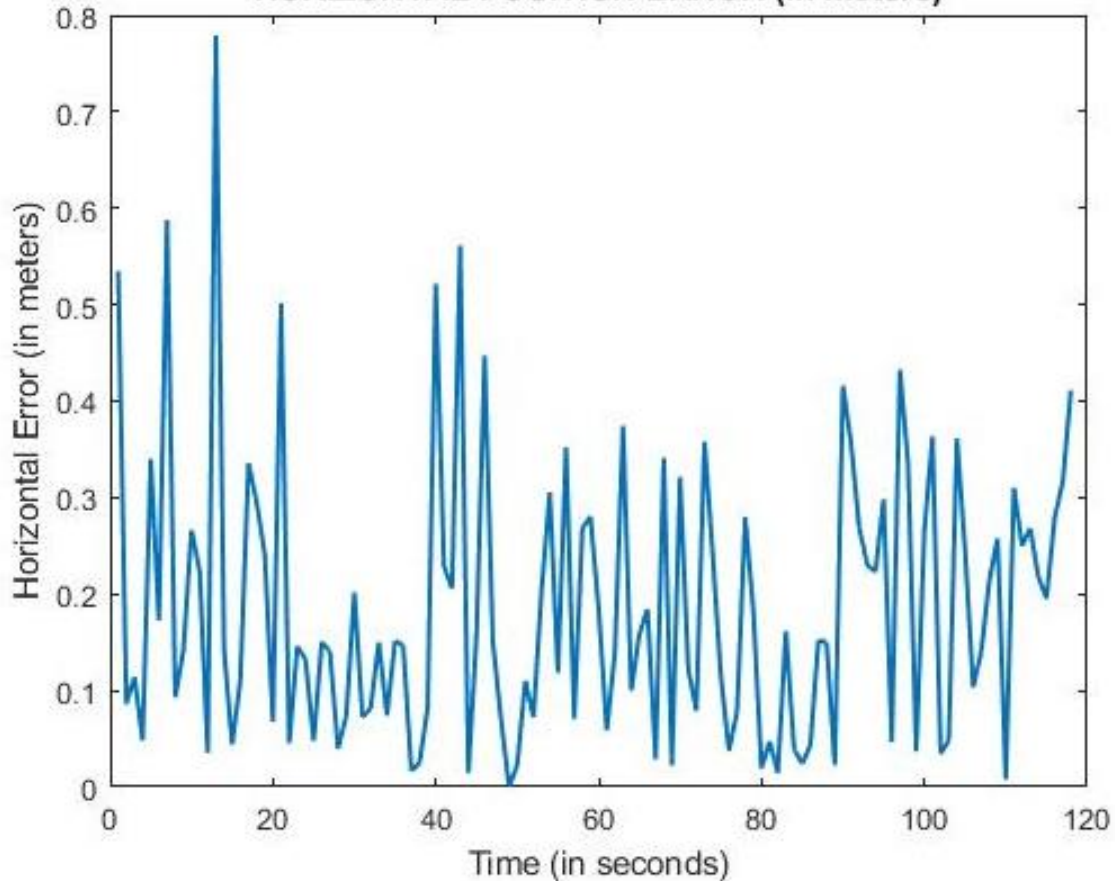
LEO satellite visibility with respect to User

## 3D Position Error: Gaussian Error with mean 0.1 & SD 0.003 m/sec

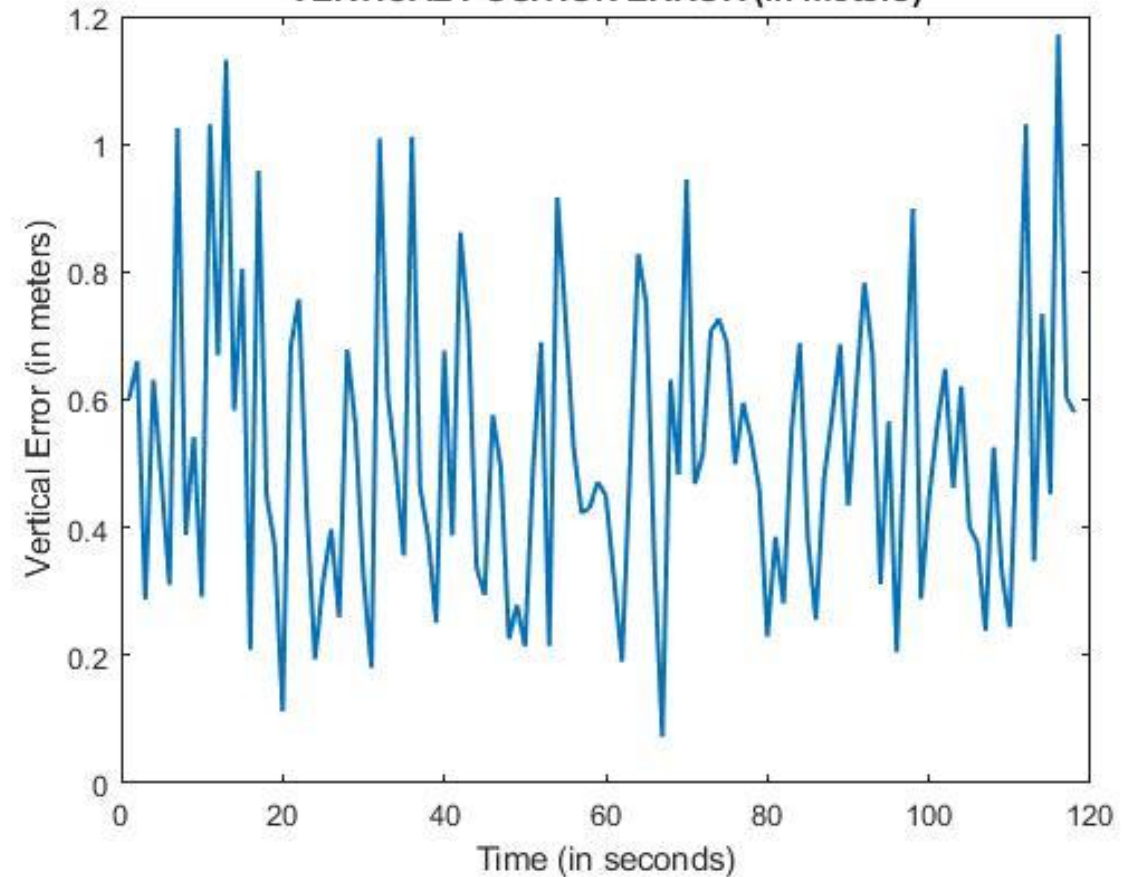


# Horizontal & Vertical Position Error

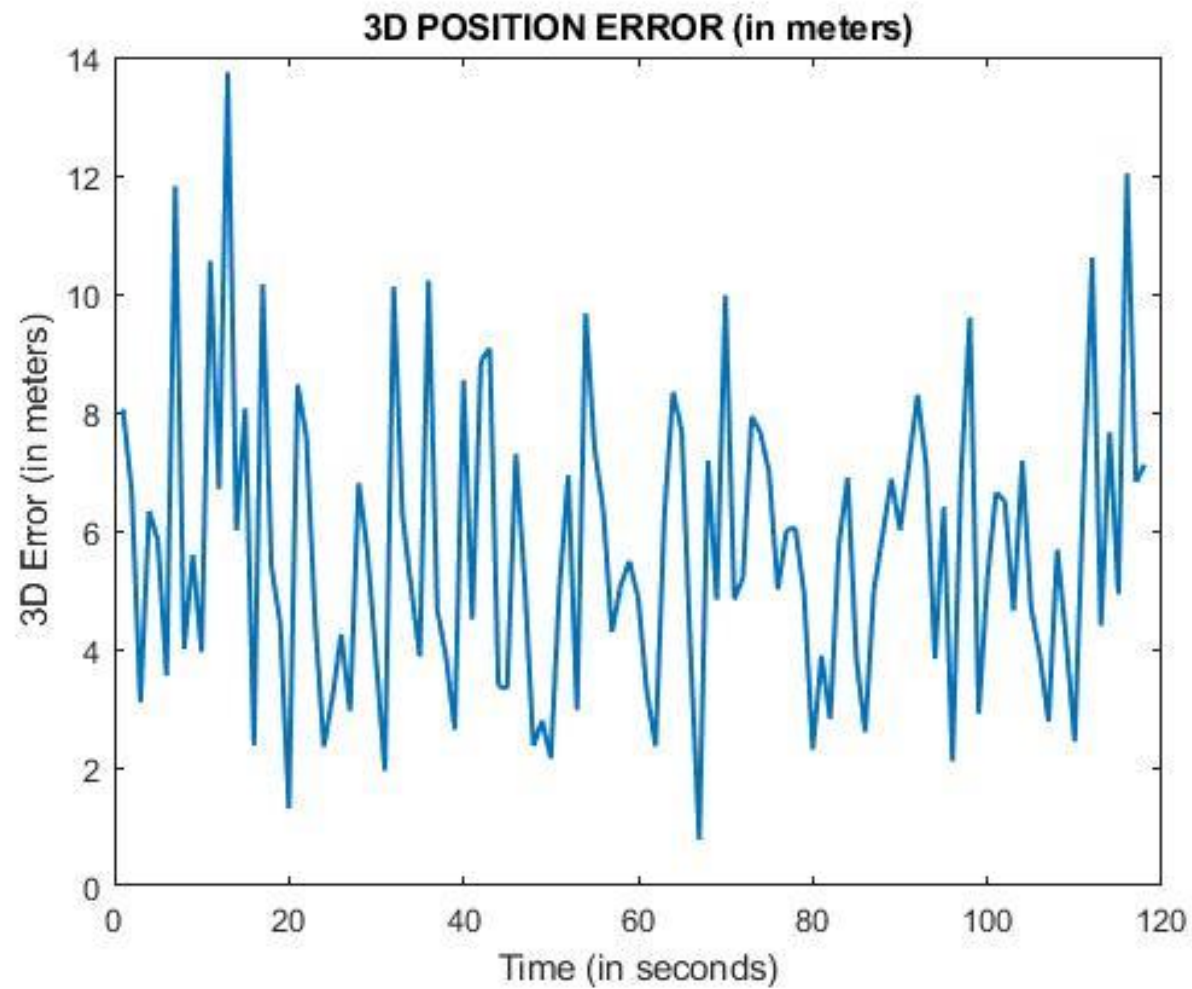
HORIZONTAL POSITION ERROR (in meters)



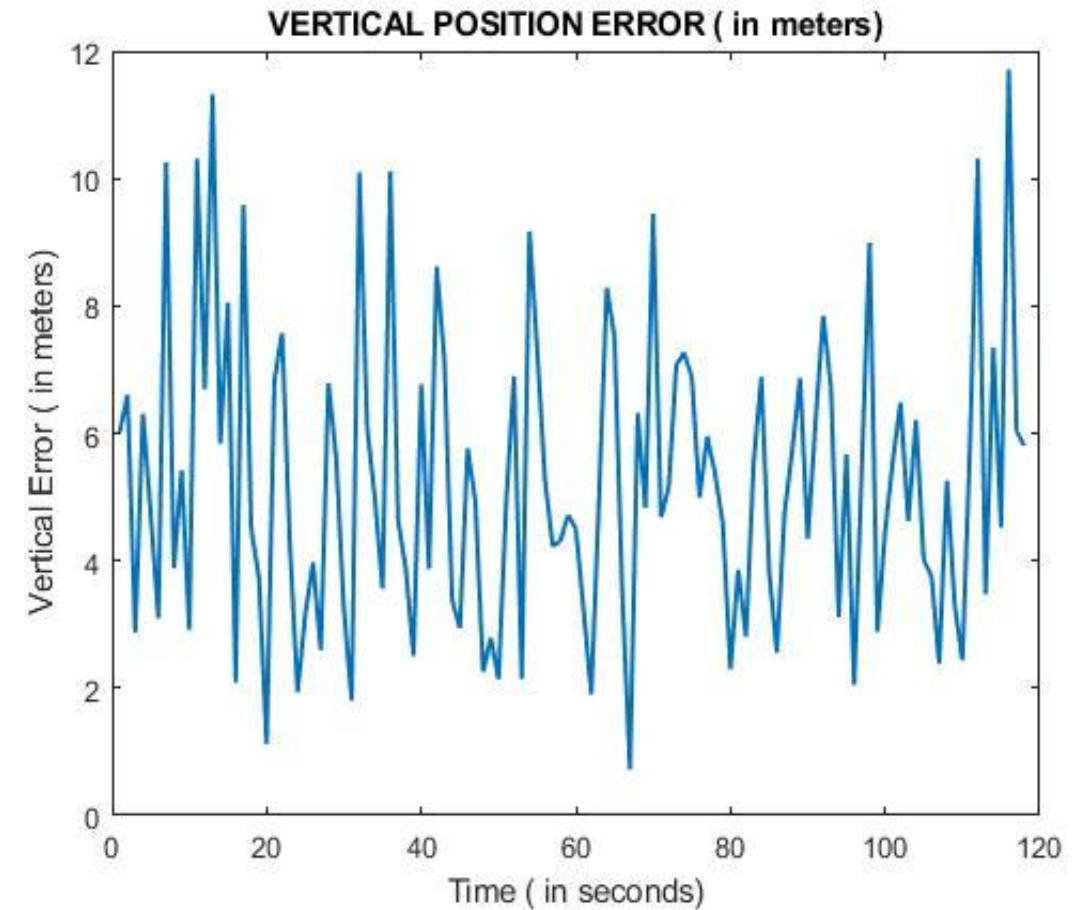
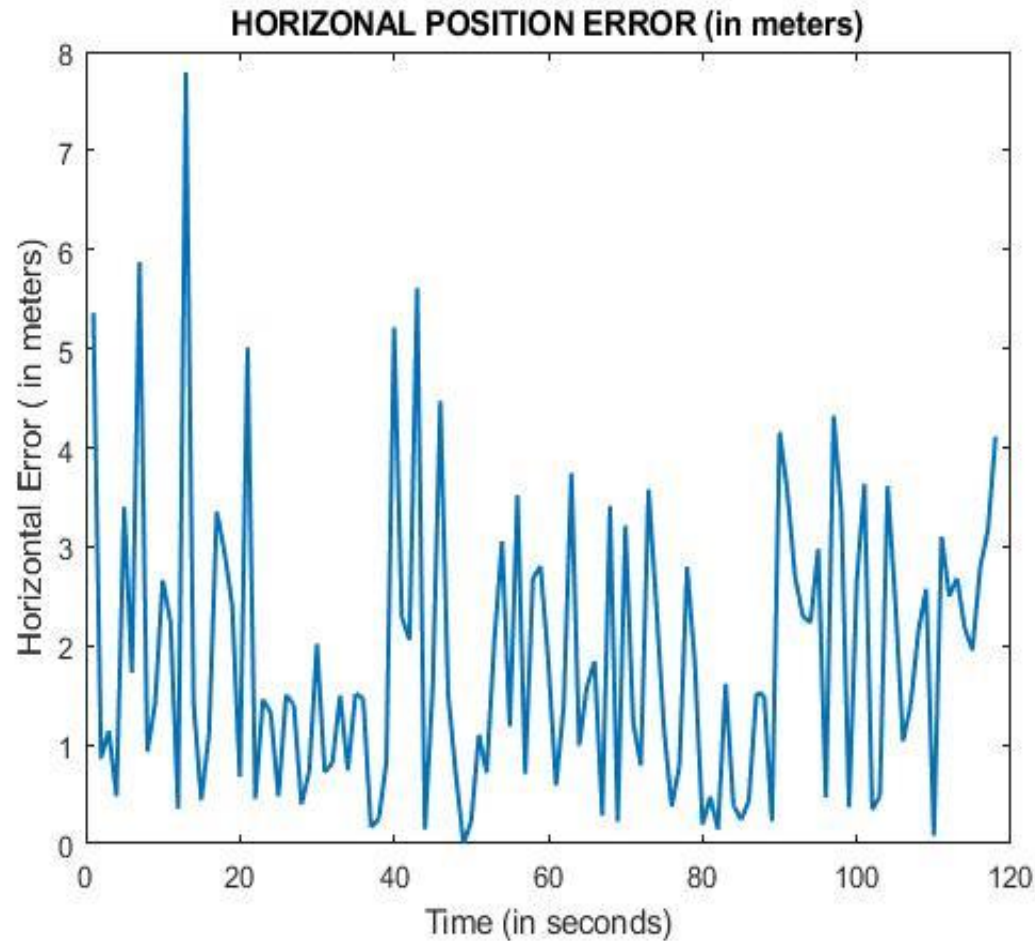
VERTICAL POSITION ERROR (in meters)



## 3D Position Error: Gaussian Error with mean 0.1 & SD 0.03 m/sec



# Horizontal & Vertical Position Error





# Conclusion

- LEO Doppler based positioning is attempted with simulated data.
- LEO 3-D, horizontal & vertical position accuracy of cm level is achieved after introducing 0.1 m/sec error in the range rate with SD as 0.003 m/sec & 0.03 m/sec.
- Simulation of actual LEO orbit and other impairments is underway with EKF Positioning approach.
- A prototype receiver hardware is being developed to demonstrate the concept.

