



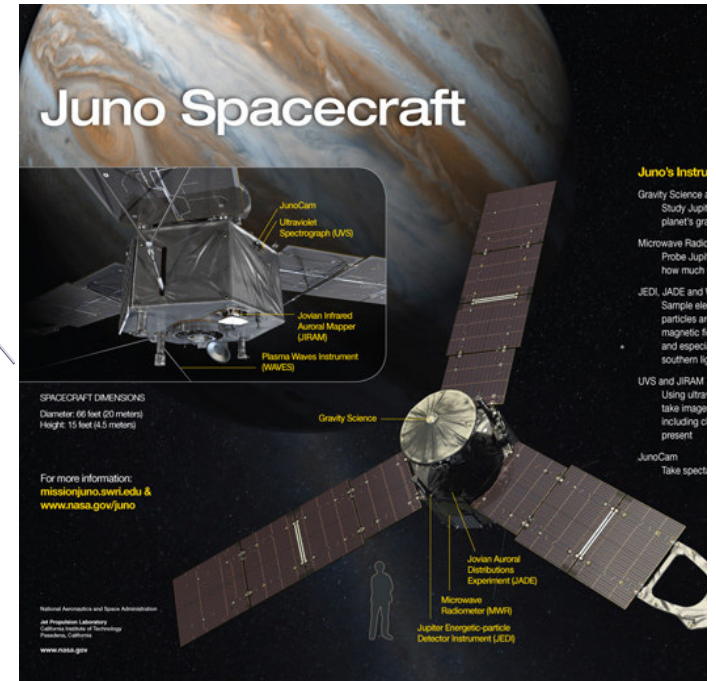
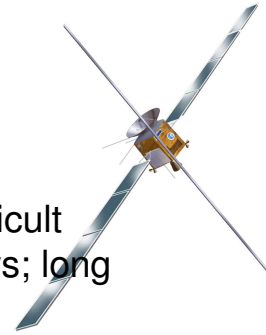
# European Space Nuclear Power Programme: UK Activities

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# Why Nuclear?

- **Solar power can be used with low risk:**
  - For orbiters of Venus/Earth/Mars
- **Solar power can be used with more difficulty:**
  - For Mercury orbiters
  - For Mars landers
- Energy budget is marginal –surviving the night is difficult
  - For cometary missions (Rosetta has 50m<sup>2</sup> arrays; long hibernations)
  - (tbc) for Jupiter missions
- *“Juno will be the first solar-powered spacecraft designed to operate at such a great distance from the sun...” [NASA website]*
- JGO/JUICE/LAPLACE is specified as solar-powered –severely constrains operations – 51m<sup>2</sup> array, lithium ion batteries for peak and eclipse power



(source: Courtesy of NASA)

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- **Nuclear power is required:**
  - For Saturn and beyond
  - For lunar night
  - For human Exploration (Reactors -EC responsibility?)
- **Nuclear power simplifies**
  - Operations at Jupiter
  - Mars landers/rovers

Current ESA plans:

- RHU
- RTG
- not Reactors

## Enabling Exploration with Small Radioisotope Power Systems



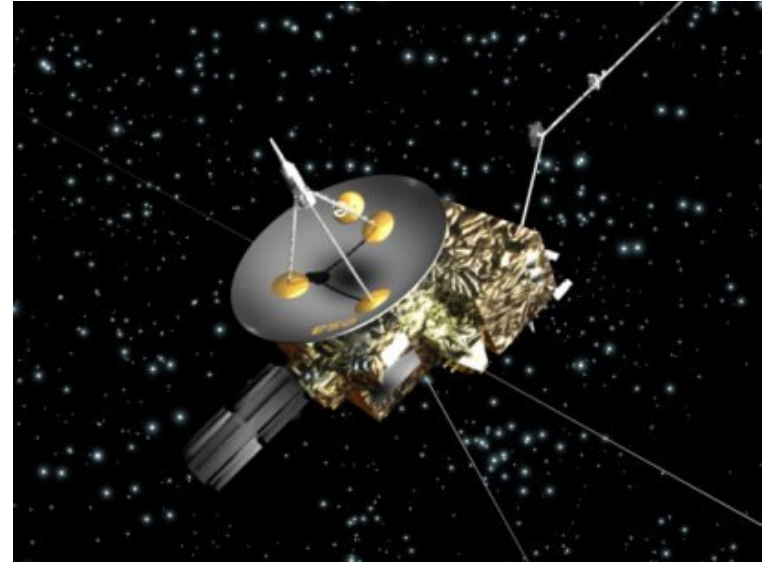
# Heritage and science return



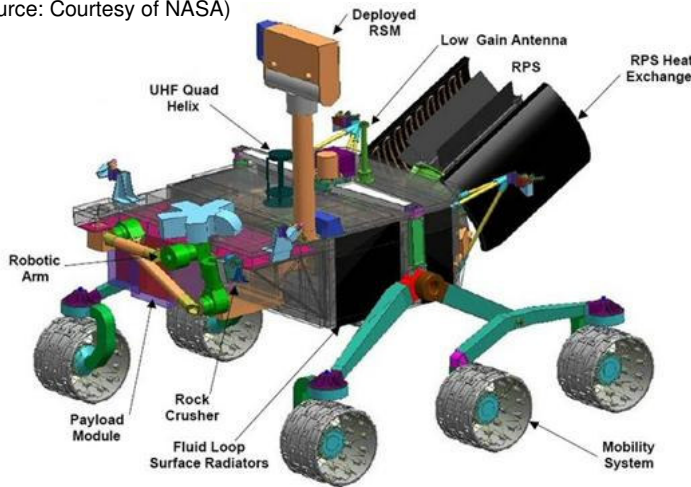
(source: Courtesy of NASA)

<Multi-Mission Radioisotope Thermoelectric Generator is planned for use on the Mars Science Laboratory mission.

>On 10 June 2009, Ulysses became the longest running ESA-operated spacecraft 18 years 246 days.



(source: Courtesy of NASA)



NASA's Mars Science Laboratory - Curiosity rover, powered by a Pu-238 fuelled radioisotope thermoelectric generator.

- Key advantage of operating continuously
- Enabling operation at vast distances from the sun.
- Independent of unavoidable variations in sunlight.
- Provide power for long periods of time (significantly longer than chemical batteries),
- Has little sensitivity to temperature, radiation, dust or other space environmental effects.
- Suited to missions involving autonomous, long-duration operations in the most extreme environments in space and on planetary surfaces.





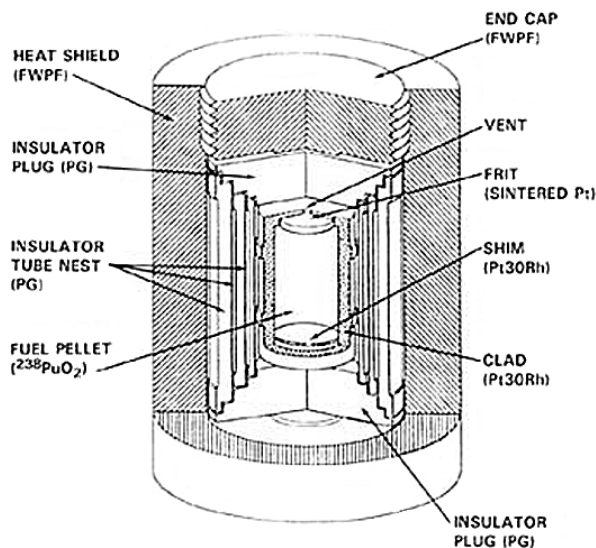
# RHU

- Hundreds flown to date (including 35 on Ulysses and 157 on Cassini-Huygens)
- Use Pu-238 as their fuel in its chemical form  $\text{PuO}_2$

Characteristic	Russian Angel RHU	US RHU
Diameter (mm)	40	26
Height (mm)	60	32
Weight (g)	200	40
$\text{PuO}_2$ Mass (g)	20	2.7
Pu mass (g)		2.38
$^{238}\text{Pu}$ mass (g)		1.91
Thermal Power (W)	8.5	1

(source: ESA Radioisotope Study Final Review – SEA Ltd)

## LIGHTWEIGHT RADIOISOTOPE HEATER UNIT



Pt = Platinum  
Rh = Rhodium  
PG = Pyrolytic Graphite

$^{238}\text{PuO}_2$  = Plutonium Dioxide  
FWPF = carbon-carbon composite woven with perpendicularly oriented graphite fiber

- Decay of radioactive material generates heat
- RHU's are small pellets of encapsulated radioactive material
- Used to keep vital spacecraft systems warm



# RTG

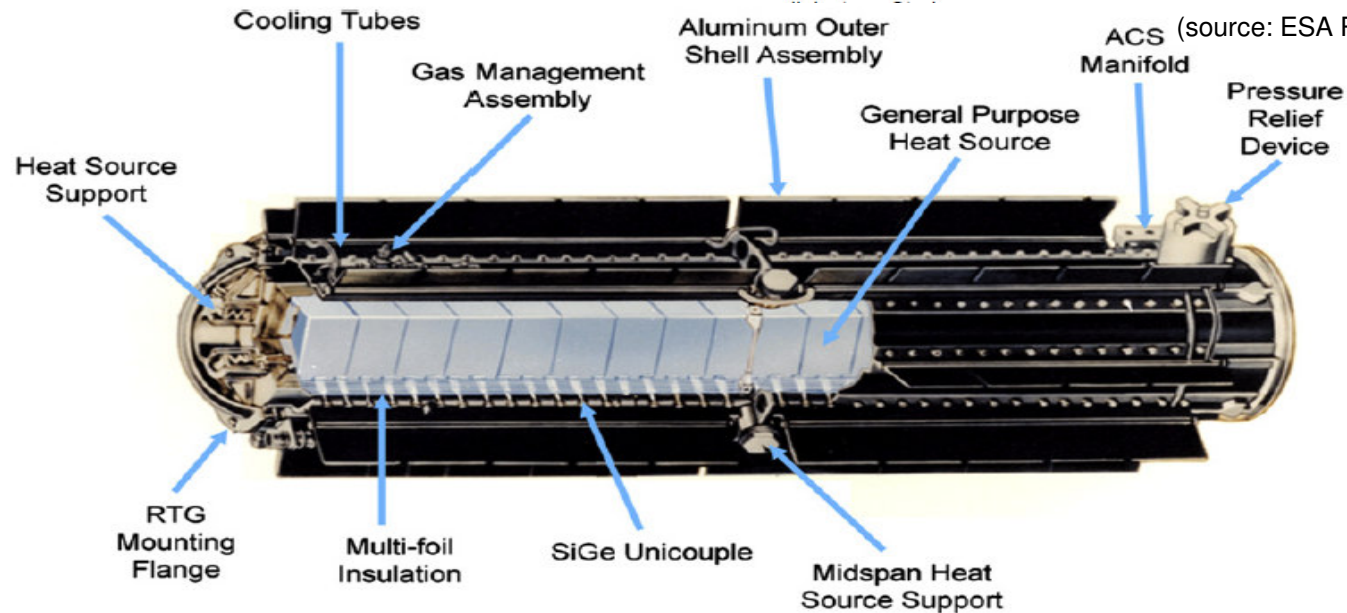
- Decay of radioactive material generates heat
- Heat used to generate electricity by thermal electric conversion

**51 m<sup>2</sup> solar panels with a mass of 282 kg producing 539 We or 1.9 We/kg \*\*.**

**As opposed to ~3We/kg to ~7 We/kg depending on the flavour of radioisotope generator.**

- 41 RTGs flown on 23 US satellites; all used Pu-238 fuel

RTG Type	Number Flown	Number of Satellites	We BOL	Pu-238 (kg)	Mass (kg)
SNAP-3B	2	2	2.7	0.2	2
SNAP-9A	2	2	25	1	12
SNAP-19	11	4	40.3	1	14
Modified SNAP-19	4	2	42.7	1	15
SNAP-27	5	5	73	3.8	20
MHW-RTG	10	4	160	4.5	38
GPHS-RTG	7	4	285	7.7	56
MMRTG	Prototype Phase		125	3.4	43
ASRG	Prototype Phase		143/160	0.9	23



(source: ESA Radioisotope Study Final Review – SEA Ltd)



# Key elements of NPS work plan 2009-2012

- Radioisotope production
  - Radioisotope selection
  - Production technique: how?, where?, cost?
  - Validation through sample programme
- Radioisotope encapsulation
  - Capsule design
  - Mechanical/thermal interface with converter
  - Prototyping (TR4), for RHU and RPS
- Conversion technique
  - Two power conversion techniques will be explored: Thermoelectric and Stirling engine.
  - Design and breadboards are foreseen in both cases
  - Thermoelectric conversion suitable for small-power generators

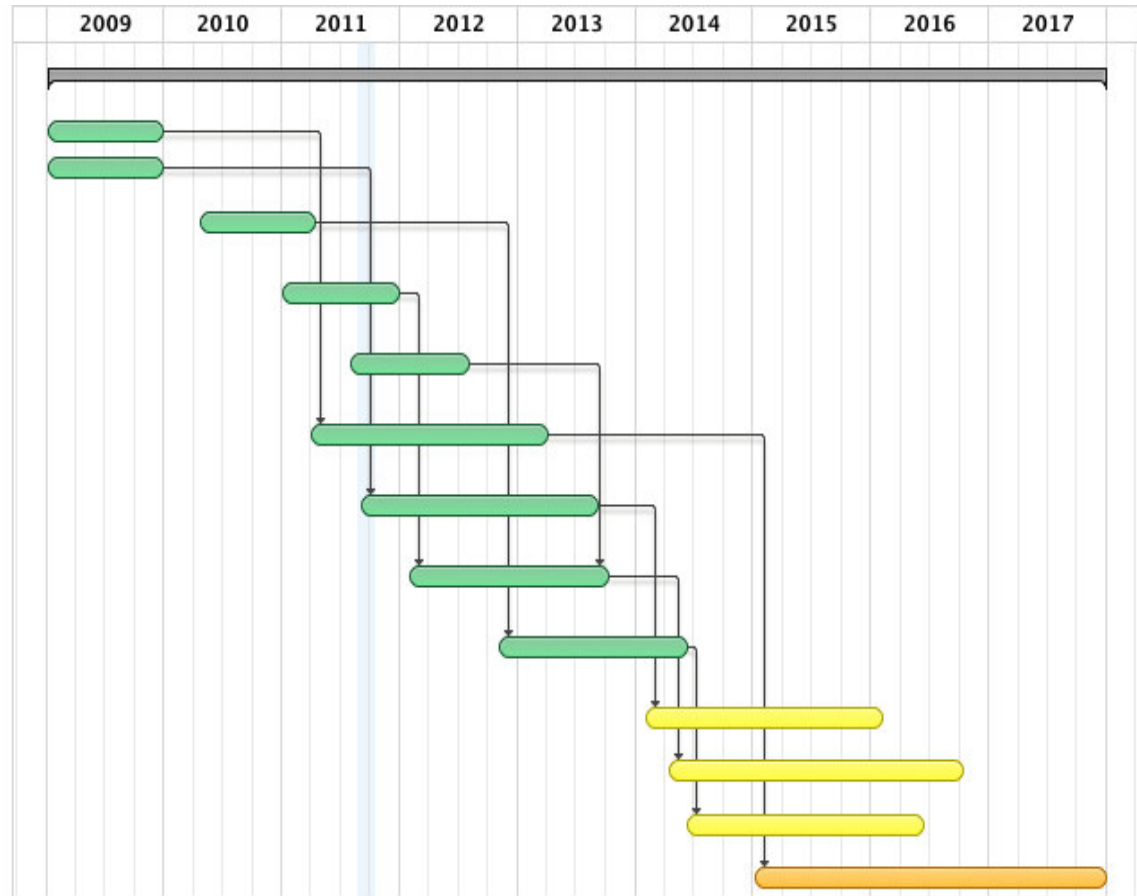
(As established by ESA Science and Robotic Exploration Advanced Studies and Technology Preparation)



# European Roadmap

Task

- 1) **Space Nuclear Power Development Programme (RTG and SRG)**
  - 1.1) Isotope Evaluation
  - 1.2) Stirling Engine Requirements
  - 1.3) Small Scale RTG Development to TRL3 (TRP)
  - 1.4) Nuclear Aeroshell and Capsule TRL2-3 (TRP)
  - 1.5) Iridium Alloy Selection and Welding (New Member States – ESA)
  - 1.6) Isotope Production Phase 1 and 2 (MREP)
  - 1.7) Stirling Engine Development Phase 1 (MREP)
  - 1.8) Nuclear Aeroshell, Capsule & Launch Safety TRL4 (MREP)
  - 1.9) Small Scale RTG Development to TRL5 (MREP)
  - 1.10) Stirling Engine Development TRL 6 (MREP)
  - 1.11) Isotope Encapsulation TRL 6 (MREP)
  - 1.12) Small Scale Prototype RTG Development TRL6 (MREP)
  - 1.13) Isotope Production Phase 3? (MREP)



# European Nuclear Power System Requirements

- **Lifetime: Up to 20 years**
- **Power:**
  - RPS 100 We ( $\sim 1538 W_{th}$  assuming 15% - 30% efficiency –typical for an RTG)
  - RHU 5  $W_{th}$
- **Radiation levels produced by the isotope must be compatible with ground personnel and with operation of spacecraft electronics**
- **Chemical form of the radioisotope used as a heat source must be:**
  - Chemically stable, even at high temperatures ( $\sim 2000$  K in case of launch failure)
  - Chemically compatible, at the likely working temperatures, with the material used as first level containment
  - Thermally stable such that it will not chemically or physically decompose at the likely working temperature (620 K for RHU, 1300 K for RTG)
  - Have a known solubility in water such that the environmental impact in the case of accidental release can be calculated
- **Must be obtainable (in Europe) and affordable**





# UK expertise

Company/University/Research Lab	ESA Programme Development Area/Expertise
Astrium UK	RTG Development, System Design
Cardiff University	Thermoelectric materials
European Thermodynamics	Thermoelectric Generators
Johnson Matthey Metals	Encapsulation & Aeroshell
Lockheed Martin, UK	Encapsulation & Aeroshell
Nanoforce Ltd	Encapsulation & Aeroshell; Thermoelectrics; Materials Processing
National Nuclear Laboratory	Radioisotope Production; Encapsulation & Aeroshell; Facilities
National Physical Laboratory	Metrology & Thermoelectric materials
Queen Mary University of London	Materials Science
Reviss UK	Radioisotopes & Transportation of Nuclear Materials
SEA	Stirling Conversion, Encapsulation & Aeroshell
RAL	Stirling Conversion
Rolls Royce	HiPER - Nuclear Electric Propulsion using reactor technologies
University of Leicester	RTG Development; Encapsulation & Aeroshell; Materials Science
University of Oxford	Stirling Conversion



# Isotope selection

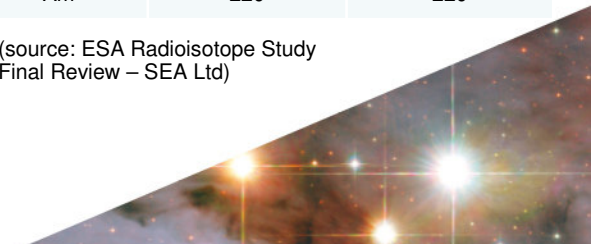
- Pu-238 has been the isotope of choice for five decades
- Proven track record
- Considerable USA and Russian investment in the technology
- Problem, sourcing is becoming an issue

## UK team recommendations:

- Not enough separated isotopes in ESA for even one 100W RTG
- UK owns enough Pu-238 for 25 of 5W RHUs – insufficient for an RTG
- Potentially attractive immediate task
- Long term recommendation is Am-241
- Quantities exist in civil PuO<sub>2</sub> stocks
- Terrestrial applications (in addition to current use in smoke alarms) include oil well logging, moisture sensors, (with Be) neutron radiography, nuclear micro-batteries and terrestrial RTGs.
- (subject to detailed study) NNL facilities considered to be capable of handling the material

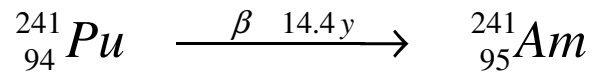
Characteristic	<sup>238</sup> PuO	<sup>241</sup> Am
Chemical form	PuO <sub>2</sub>	Am <sub>2</sub> O <sub>2</sub>
Mass of compound for RTG (g)	3747	14700
Power (W/Kg)	0.411	0.105
Theoretical density (g/cm <sup>3</sup> )	11.5	12.7
Total volume (cm <sup>3</sup> )	326	1158.3
Power (W/cm <sup>3</sup> )	4.72	1.3
mp (K)	2673	2478
Thermal Con. (W/mK) <sub>600K</sub>	6	1
Thermal Con. (W/mK) <sub>1300K</sub>	3	1
Fabrication	ox	ox
Special handling	Box	Box
Solubility pH- [M] <sub>TOT</sub> (M)	10 <sup>-9</sup>	10 <sup>-6</sup> to 10 <sup>-4</sup>
Isotope	Power output after 15 years (W <sub>m</sub> )	Power output after 20 years (W <sub>m</sub> )
<sup>238</sup> Pu	1010	990
<sup>241</sup> Am	220	220

(source: ESA Radioisotope Study Final Review – SEA Ltd)



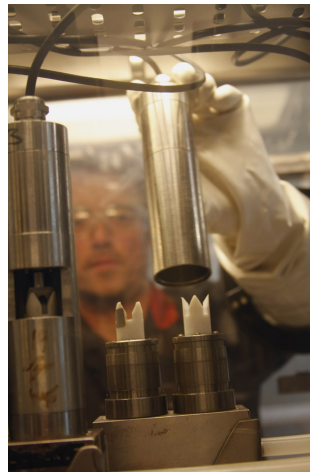
# Am-241

- 100% isotopically pure Am-241 forms in separated civil plutonium from the decay of Pu-238



- Large quantities of 20+ year stored civil plutonium exists
- Aged civil PuO<sub>2</sub> contains 25-50g per kilo
- No need for irradiation in a reactor
- Availability of Am-241 can compensate for other limitations: mass and energy conversion between Pu-238 is not linear.

- NNL has a full range of facilities for handling radioisotopes including actinides such as Americium and plutonium.
- A study is currently being undertaken on behalf of ESA to provide a conceptual design and costing for a plant to separate americium from civil plutonium.
- It is planned to separate the first Am-241 from civil plutonium at the NNLs facilities next FY using the flow sheet proposed for the full scale plant
- NNL have full ISO 9001/14001 accreditation for its facilities.



Centrifugal contactor system  
(<sup>238</sup>Pu → <sup>241</sup>Am separation)



P2 line







# System Specifics

- **Containment compliance (launch safety)**
  - No specific ESA or UK industrial activities
  - Well understood to be a very large activity – Planning initiated
  - Expected to be led by France (Ariane5, Kourou)
- **Encapsulation and Aeroshell**
  - Study well under way
  - Issues understood
  - UK has the capabilities as do other European states
- **Thermal-Electrical conversion**
  - ESA is pursuing two routes
    - Seebeck effect
      - 2 studies in progress
    - Stirling engine
      - Requirements study complete
      - Breadboard implementation -Bids under evaluation



# Summary

- **Programme proceeding along the ESA roadmap**
  - Almost all planned activities under way
- **UK has all the technical expertise and capabilities required**
- **We welcome interest and collaboration from any other states.**

