QZSS

The Japanese Quasi-Zenith Satellite System

Program Updates and Current status

Fifth International Committee on GNSS
Torino, Italy
October 18, 2010
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QZSS outline

Quasi-Zenith Satellite System

- **Coverage**: East Asia and Oceania region

- **Six Signals**:
  - L1C/A, L1C, L2C and L5
    - can provide seamless PNT services by combining usage with GPS.
    - Increasing coverage and availability of PNT services even in downtown and mountainous areas.
  - L1-SAI on 1575.42 MHz
  - LEX on 1278.75 MHz
    - can enhance GPS performance by transmitting error correction and integrity information.

- **Accelerate**: the modernization of GPS in Asia Oceania region.

- **Platform**: for Multi-GNSS augmentation.

- **First satellite**: launched in September 2010.

- **Plan**: the 2nd and 3rd satellite will be approved after assessment of the technical and application demonstration result.
QZSS Service
Ground Track of a QZSS Satellite

At least one QZSS satellite can be observed more than specified elevation angle any time. For instance, users in orange colored area can receive at least one QZSS satellite with 70 degrees or more.
QZSS Service Coverage
the Number of Satellites

Southeast Asia can observe 3 QZSS satellites almost of time.

Most East Asia and Oceania countries can observe more than 2 QZSS satellites every time.
## QZSS Service

### QZSS Signals

<table>
<thead>
<tr>
<th>Signal</th>
<th>Frequency</th>
<th>Notes</th>
</tr>
</thead>
<tbody>
<tr>
<td>L1-C/A</td>
<td>1575.42MHz</td>
<td>Complete compatibility and interoperability with existing and future modernized GPS signals</td>
</tr>
<tr>
<td>L1C</td>
<td>1575.42MHz</td>
<td>Differential Correction data, Integrity flag, Ionospheric correction</td>
</tr>
<tr>
<td>L2C</td>
<td>1227.6MHz</td>
<td>Almanac &amp; Health for other GNSS SVs</td>
</tr>
<tr>
<td>L5</td>
<td>1176.45MHz</td>
<td>Compatibility with GPS-SBAS</td>
</tr>
<tr>
<td>L1-SAIF*</td>
<td>1575.42MHz</td>
<td>Compatibility with GPS-SBAS</td>
</tr>
<tr>
<td>LEX</td>
<td>1278.75MHz</td>
<td>Experimental Signal with higher data rate message (2Kbps)</td>
</tr>
<tr>
<td></td>
<td></td>
<td>Compatibility &amp; interoperability with Galileo E6 signal</td>
</tr>
</tbody>
</table>

* L1-SAIF: L1-Submeter-class Augmentation with Integrity Function
QZSS Service

Time System and Frame

- **Time Reference System: QZSST**
  - The length of one second is identical to *International Atomic Time (TAI)*.
  - Integer second offset for TAI is the same as GPS, and TAI is **19 seconds** ahead of QZSST.
  - Interface with GPS:
    - The SV clocks of QZS and GPS satellites are both controlled with respect to the offset with the GPS time scale (GPST).
    - **GQTO**: The time scale offset with the GPS is less than **2.0 [m]** (95%).

- **Geodetic Reference Frame: JGS**
  - The QZSS coordinate system is known as the *Japan satellite navigation Geodetic System (JGS)*. This coordinate System is operated so as to approach the *International Terrestrial Reference System (ITRS)*.
  - The coordinate system offset with GPS is less than **0.02 [m]**.
QZSS Service

Accuracy

- **The Signal-in-Space (SIS) User Range Error**
  - is less than 1.6 m (95%) without time and coordination offset error.
- **User Positioning Accuracy**
  - define as positioning accuracy combined GPS L1_C/A and QZSS L1_C/A for single frequency user, L1-L2 for dual frequency user.

<table>
<thead>
<tr>
<th></th>
<th>Specification</th>
<th>Simulation result</th>
</tr>
</thead>
<tbody>
<tr>
<td>SIS-URE</td>
<td>1.6m (95%)</td>
<td>1.5m (95%)</td>
</tr>
<tr>
<td>Positioning Accuracy</td>
<td>21.9m(95%)</td>
<td>7.02m(95%)</td>
</tr>
<tr>
<td>Positioning Accuracy</td>
<td>7.5m (95%)</td>
<td>6.11m(95%)</td>
</tr>
<tr>
<td>Single frequency user</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Dual frequency user</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

- **L1-SAIF signal can provide WDGPS correction data, its positioning accuracy is 1m (1 sigma rms) except in cases of large multipath error and large ionospheric disturbance.**
QZSS System Description
# QZSS System Description

## the Satellite named “Michibiki”

<table>
<thead>
<tr>
<th>Mass</th>
<th>Approx. 1,800kg (dry) (NAV Payload: Approx. 320kg)</th>
</tr>
</thead>
<tbody>
<tr>
<td>Power</td>
<td>Approx. 5.3 kW (EOL) (NAV Payload: Approx. 1.9kW)</td>
</tr>
<tr>
<td>Design Life</td>
<td>10 years</td>
</tr>
</tbody>
</table>

- **Radiation Cooled TWT**
- **L1-SAIF Antenna**
- **TWSTFT Antenna**
- **C-band TTC Antenna**
- **Laser Reflector**
- **L-band Helical Array Antenna**

The satellite has a mass of approximately 1,800kg (dry) with a NAV payload of approximately 320kg. The power requirement is approximately 5.3 kW at end-of-life (EOL) with a NAV payload of approximately 1.9 kW. The design life of the satellite is 10 years.
QZSS System Description

the Navigation Payload

- Rb Atomic Clock
- Time Keeping Unit
- Synthesizer
- Navigation Onboard Computer
- Modulator
- Amplifier
- MUX

TT&C Subsystem

Navigation Signal
Sine Wave
Baseband Signal (Navigation Message + PRN Code)
Signal of Two Way Satellite Time and Frequency Transfer

Uploaded Data (including Remote Synchronization Signal (by AIST))

Laser Reflector

Ku-Ant

L1-SAIF ANT

JAXA

NICT
(previously: CRL)

2010/10/18

5th International Committee on GNSS: Turin, Italy
Okinawa is primary TT&C station for nominal operation.
LEOP operation is to be conducted by using JAXA’s Ground TT&C Network.
Development, Launch & Next Step
Development

the MCS and the TTC Station

- **Master Control Station (MCS)**
  - MCS is located in Tsukuba Space Center
  - Installation and integration test were completed
- **Tracking & Control Station**
  - Built new C-band TT&C capability in Okinawa
  - Installation and integration test were completed
Development of the Monitor Stations

- 9 sites distributed in Asia Oceania region
- Established in collaboration with international partners
  - Guam;
    - NOAA, National Weather Service Forecast Office (WFO)
  - Hawaii;
    - NASA, Kokee Park Geophysical Observatory (KPGO)
  - Bangalore;
    - Indian Space Research Organization (ISRO)
  - Canberra;
    - Geosciences Australia (GA)
  - Bangkok;
    - Asian Institute of Technology (AIT)
Development

the Payload Development

PFM L-band Antenna Pattern Test
(July 2008)

NAV Payload PFM TVT
(Jan 2009)
Development
the Satellite Development


Initial Alignment Test (9/5/2009-9/10/2009)


Sine Vibration Test (7/12/2009-6/1/2010)

Separation Shock Test (1.13/2010-1.14/2010)

TVT (10/1/2009-10/30/2009)

Ship to Launch site (6/5/2010)
...and the Launch

Sep.11 at 20:17
Lift-Off  @JAXA TNSC

Sep.11 at 20:45
Separation from H-IIA
Launch and Next Step

Launch and OOC

• **Launch:**

  The 1st satellite (QZS-1; MICHIBIKI) was launched on September 11th, 2010, JST.

• **On-Orbit-Checkout: OOC:**

  • 3 months On-Orbit-Checkout (OOC) period after the launch;
    • Orbit raising in the Orbit Transfer Period
    • Orbit settlement in the Orbit Drift Period
    • and the Initial Check Operations.
Launch and Next Step

Next Step: Demonstrations

Demonstrations will be started after OOC

- **Technical Demonstration:**
  - Various parameters are tuned
  - Ephemeris accuracy is to be improved.
  - And after the time when SIS URE is verified to meet the specification, QZS-1 will be set healthy.

- **Application Demonstration:**
  - 101 companies will participate.
  - 58 programs will be implemented.
Organization
Organization of QZSS Development & Utilization

Cabinet

- Ministry of Internal Affairs and Communications (MIC)
- Ministry of Education, Culture, Sports, Science and Technology (MEXT)
- Ministry of Economy, Trade and Industry (METI)
- Ministry of Land, Infrastructure, Transport and Tourism (MLIT)
- Ministry of Foreign affairs (MOFA)

Ministry

Agency / Institute

- National Institute of Information and Communications Technology (NICT)
- Japan Aerospace Exploration Agency (JAXA)
- National Institute of Advanced Industrial Science and Technology (AIST)
- Geospatial Information Authority of Japan (GSI)
- Electronic Navigation Research Institute (ENRI)
- Satellite Positioning Research and Application Center (SPAC)
Service Provided and Provision Policies
QZSS Service Provision Policies

- GPS interoperable signals, L1 C/A, L2C, L5 and L1C, are to be provided on the basis of no direct user fee.

- GPS performance enhancement signals, L1-SAIF and LEX, charging policy is under examination.
Inter-GNSS
Inter-GNSS

Views on Interoperability

• Interoperability is NOT mandatory, but highly desirable in the users’ point of view.

• Japan will continue to keep the interoperability at the higher level for all user communities including low-cost receivers
  • L1 and L5 with GPS, Galileo, COMPASS, as well as future GLONASS CDMA signals
  • L2C with GPS
  • LEX with Galileo
Inter-GNSS

Views on Compatibility

- Compatibility is a mandatory requirement to share the same frequency bands among multi GNSS systems without harmful interference.

- Japan will continue to comply with the international rule and consensus.
  - ITU Radio Regulation
  - ICG definition
Multi-GNSS
Multi-GNSS

Total Number of GNSS Satellites

We will have over 100 GNSS satellites in this decade.
Multi-GNSS

New Applications are Expected

- User benefits from Multi GNSS
  - Increase in usable SVs, signals and frequencies
  - Increase in availability and coverage
  - More robust and reliable services
  - Higher accuracy in bad conditions
  - Less expensive high-end services

Emerging new and expanding existing applications are to be expected.
Multi-GNSS

Asia Oceania Showcase

Visible Satellite Number (mask angle 30 deg)

Asian People can use multi-GNSS signals earlier than other region in the world

- Multi-GNSS
- More Stars, Signals
- Multi-Frequency
  
- Higher accuracy
- More reliable, robustness
- Wider coverage

New Applications
Multi-GNSS

Demonstration Campaign

- *is a series of activities for five years from 2010*
- *comprises the following parts*

1. **Multi-GNSS Monitoring Network**
   - CORS (Continuously Operating Reference System), *Data center, Analysis Center*
   - *Sharing resources and observed data among participated organization*

2. **Applications Development & Demonstration**
   - *Development of multi GNSS applications*
   - *Carrying out Experiments and Demonstrations*

3. **Regional Work Shop**
   - *Annual base Workshop in Asia Oceania region*
   - *Announcement of joint experiment plans and reporting results of the experiments*
Multi-GNSS Demonstration Campaign

Planned Applications

Monitoring Network

Application Demonstration

Disaster Mitigation

Precise Positioning

ITS, Mapping, LBS

Other, ionospheric observation etc

Regional Workshop

1st Asia Oceania Regional Workshop on GNSS, 25, 26 JAN, 2010, Bangkok

195 Participants, 18 Countries, 95 Organizations
Next is 21, 22 Nov. 2010 @ Melbourne, Australia

http://www.multignss.asia/
Multi-GNSS Demonstration Campaign
the Status (past)

• The concept of “Multi-GNSS demo. Campaign” was presented at ICG-4 and obtained endorsements from Two WGs (WG-A and D)
• The 1st regional Workshop was held at Bangkok, Thailand on January 25-26, 2010 successfully.
  • Hosted by SPAC*, JAXA, and GISTDA** of Thailand and supported by UN International Committee on GNSS (ICG)

• 195 participants from 18 countries
• 4 discussion groups were established for future joint experiments
  1. Multi-GNSS monitoring network
  2. Disaster Mitigation and Management
  3. Precise Positioning
  4. ITS

*SPAC: Satellite Positioning Applications Center
** GISTDA: Geo-Informatics and Space Technology Agency

http://www.multignss.asia/
Multi-GNSS Demonstration Campaign

-the Status (future)-

- JAXA started a procurement for multi-GNSS receivers for the multi-GNSS monitoring network. - The 2nd Workshop is to be held in Melbourne, Australia on November 21-22, 2010.  
  - To discuss future joint demonstration projects  
     - Multi-GNSS Monitoring NW  
     - Disaster Mitigation  
     - Precise Positioning  
     - Intelligent Transportation System (ITS)  
     - Mapping, Location Based System
Summary

- **QZSS** is a regional space-based PNT system covers East Asia and Oceania region and transmits six civil PNT signals.
- **First satellite (QZS-1; MICHIBIKI)** was launched on September 11th, 2010, JST.
- **We are performing 3-month-On-Orbit-Checkout (OOC) after the launch. Technical and application demonstrations will be started after OOC.**
- **Asia Oceania is the Showcase of New Multi GNSS Era**
  - Next workshop will be held in 21, 22 Nov 2010 in Melbourne, Australia
Our Planet from QZS-1

Thank you for your attention
Backup Sheets
Political Background
Launch and Next Step

Background Act; Geo-Spatial Information

- **Basic Act on Promotion of Utilization of Geographical Information (AUGI)**
  - August 2007, Basic Act on AUGI entered into force
  - April 2008, Basic Plan for AUGI was approved by the Cabinet (based on Article 9 of the Basic Act on AUGI)
    - QZSS is a key element and implemented by phased approach;
      - Phase 1: First satellite launch and technology and application demonstration
      - Phase 2: 2nd and 3rd satellite will be launched after assessment of the result of phase 1

- **Public-Private-Partnership for Promoting Utilization**
  - QZSS project is based on the collaboration between private sector and government
Organization of 1st satellite Development

- **JAXA (MEXT)**: Development of satellite navigation technologies, integration of QZSS navigation system, and development of the first QZSS satellite and satellite control & tracking ground station.
- **NICT (MIC)**: Development of timing control technologies.
- **ENRI (MLIT)**: Development of submeter class GPS Augmentation technologies with L1-SAIF(Submeter-class Augmentation with Integrity Function) signal.
- **GSI (MLIT)**: Development of centimeter or decimeter class GPS Augmentation technologies.
- **AIST (METI)**: Development of precise onboard clock timing control technology.
Launch and Next Step

Background Act; **Space**

- **Aerospace Basic Act**
  - *May 2008, Aerospace Basic Act was enacted*
  - *June 2009, Basic Plan for Space Policy was decided by the Strategic Headquarters for Space Policy*
    
    *(chaired by the Prime Minister, based on Article 25 of the Aerospace Basic Act)*
  - **6 pillars for space activity**
    
    *(high QOL, security, diplomacy, advanced R&D, industry and environment)*

- **Space-based PNT in the Basic Plan for Space Policy**
  - *Promote highly accurate system such as QZSS and MSAS*
  - *Create new services in our daily life such as personal navigation systems, with private sector*
Launch and Next Step

Next Step: 2\textsuperscript{nd} Step

- On 27/AUG, Strategic Headquarter for Space Policy of the cabinet has released their decision on “the near term promotion of space policy”.
  - In the decision, the related Ministries have to derive the solution to the 2\textsuperscript{nd} step of QZSS together under the coordination of the cabinet secretary by the early in the next FY2011.
- The QZSS Project Team composed of the Parliamentary Secretary of the main 8 Ministries was established.
  - And on 7/SEP, the PT has started the study how to step into the 2\textsuperscript{nd} step: the 2\textsuperscript{nd} and 3\textsuperscript{rd} satellites.
- On 16/SEP, the Strategic Headquarter released the documents which relates the evidence of the FY2011 budget requirement for the 2\textsuperscript{nd} QZSS study.
- Until the end of OCT, the GNSS Expert Working Group will be also established under the committee of the Strategic Headquarter.
QZSS Connection

- Dissemination of the achievement
QZ-Vision Connects people to GNSS
Children to QZSS

http://qz-vision.jaxa.jp/
QZ-Vision Connects people to GNSS
PC to QZSS

http://qz-vision.jaxa.jp/
QZ-Vision Connects people to GNSS
Smart Phone to QZSS

http://qz-vision.jaxa.jp/
QZ-Vision Connects people to GNSS
Expert to QZSS

http://qz-vision.jaxa.jp/

Interface Document

Almanac
Ephemeris
Precise Products
Orbit de-confliction with GSO and re-orbit procedure
Avoidance of physical interference to GSO & re-orbit after operation

- Each satellite orbit has slight eccentricity so that can keep appropriate separation away from GSO. The vector of eccentricity will be maintained this appropriate separation more than 50 km during operation.

- After whole mission life, the satellite will be injected into “Disposal Orbit”, which defined as the orbit with 1000 km higher perigee altitude from GSO’s.