

experiments in IHY frame

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HISTORY AND FUTURE

- **CORONAS-I** (launched on March, 1994)
- INTERBALL-1(launched on August, 1995)
- **INTERBALL-2(launched on August, 1996)**
- CORONAS-F" (launched on July, 2001).



- **CORONAS-PHOTON** mission under development
- **RESONANCE** mission under development
- **INTERHELIOPROBE** mission under investigation

INTERBALL MISSION www.iki.rssi.ru/interball/

Interball-2 Magion-5

Interball-1

Magion-4



INTERBALL-1 Launch – 08.1995 год •Perigee - 785 км •Apogee - 200 000 км •Period - 92 ч •Inclination- 62,8 Reentry 16.10. 2000r. **INTERBALL-2-**Launch – 08.1996 год •Perigee - 770 км •Apogee - 20 000 км •Period - 6ч •Inclination- 62,8 Lost 02, 99

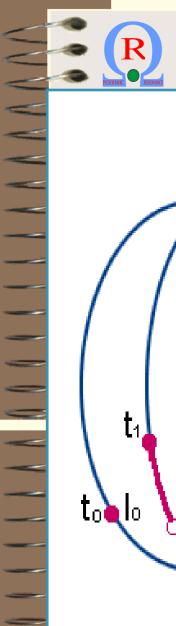




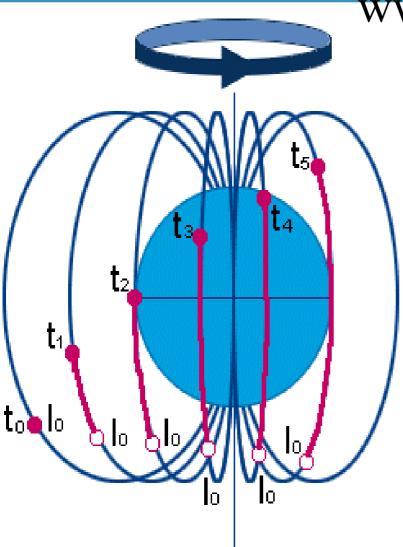
ЖИ

www.iki.rssi.ru/resonance

The space project is directed to the study of wave-particle interactions and plasma dynamics in the inner magnetosphere. Phase B of the project is supported by Federal Space Agency RUSSIA, FRANCE, UKRAINE, POLAND...



RESONANCE project



www.iki.rssi.ru/resonance

ЖИ

Magnetosynhronous

<u>orbit</u>

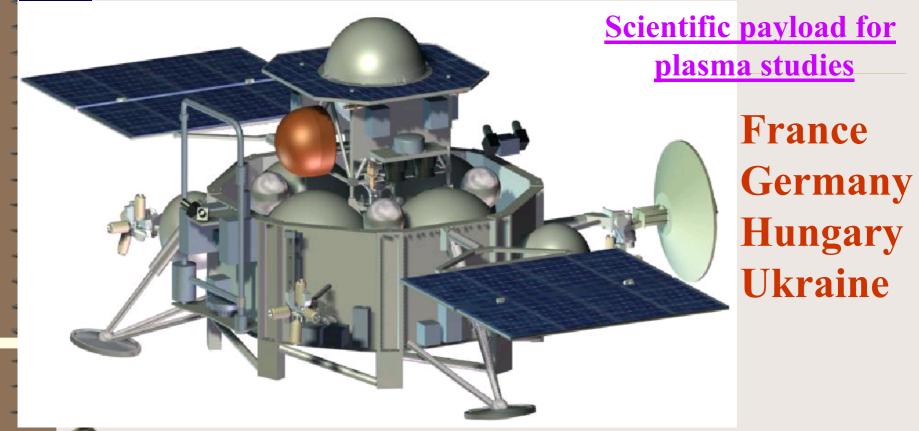
RESONANCE satellite motion along the selected magnetic flux tube mapped of the heating station. Footprint of the selected tube will be conjugate to the ionosphere above the HF heating facility HAARP.

<u>Satellite spends more then 3</u> <u>hours in the same flux tube</u> <u>(L=5.5±0.15)</u>



PHOBOS --SOIL



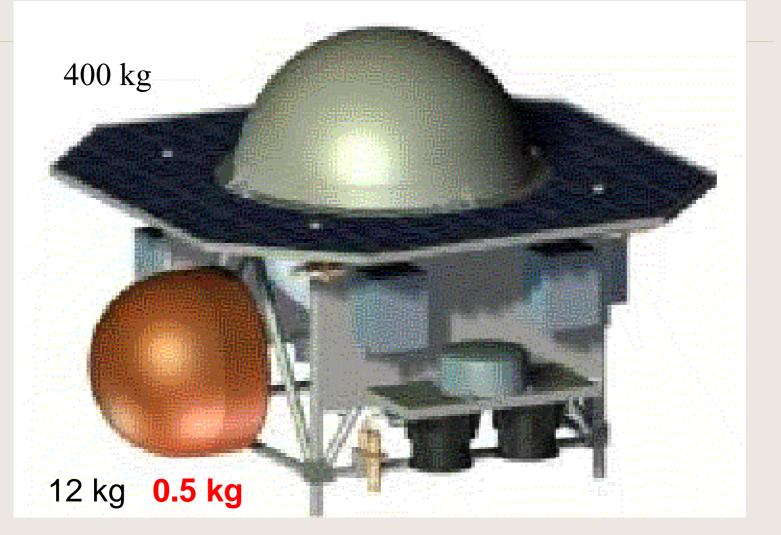


Phobos investigation (regolith, origin and evolution of the Martian moons)Martian environment conditions (dust, plasma, radiation)

-Monitoring of the Martian atmosphere and surface global dynamics

PHOBOS --SOIL *Phobos regolith sample return*







PHOBOS --SOIL

КИ

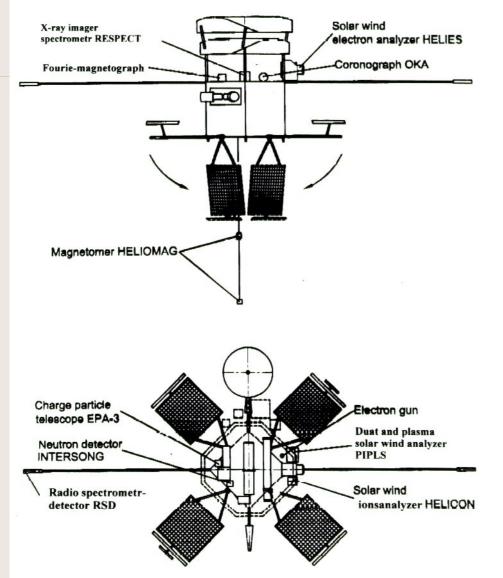
Creation on the base of platform "Phobos- soil" of the space complex

"Intergelioprobe"

for heliophysical and gravitational studies in the nearest environment of the Sun.



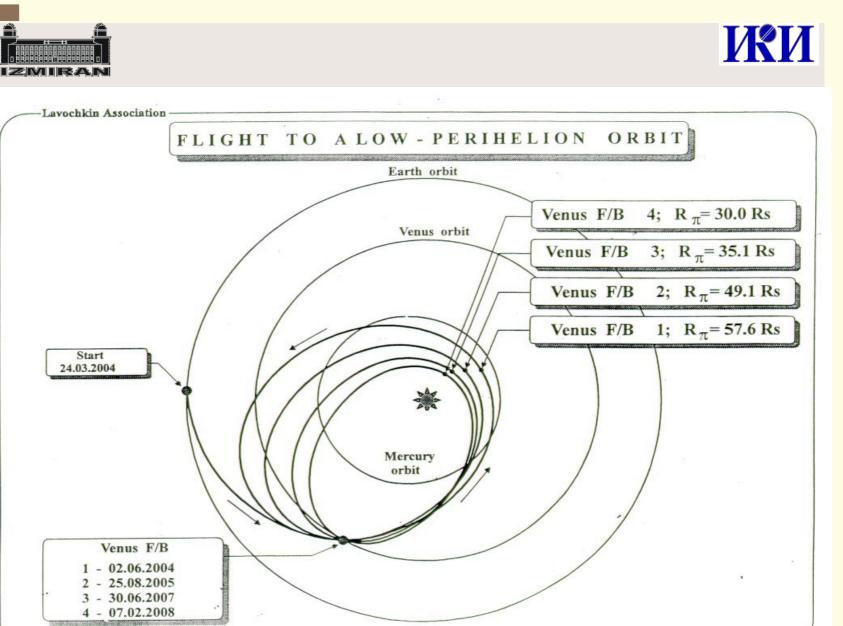
INTERHELIOPROBE SPACECRAFT



-Mass – 50-70 kg -Power – 100W -Telemetry – 60 kb/s

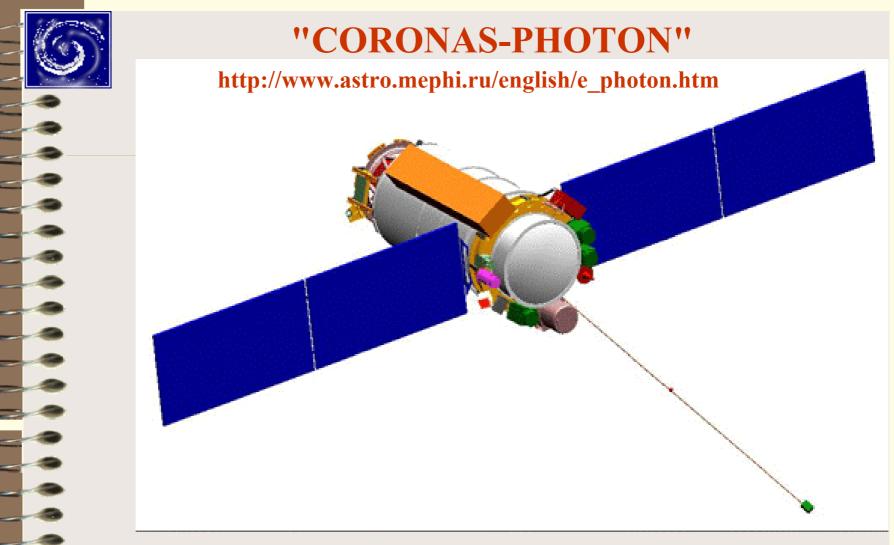
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<u>Two instrument packages</u> -Heliospheric instrumentation (in-situ measurements). -Solar instrumentation (remote sensing).



The Scientific & Technical Subcommittee of the COPUOS, 43st session, Vienna 20 February – 3 March 2006

Ballistic Department



Moscow Engineering Physics Institute (State University) MEPhI



"CORONAS-PHOTON" http://www.astro.mephi.ru/english/e_photon.htm Is the third mission in the satellite series of the "Coronas" project (two others are "Coronas-I" and "Coronas-F"). Goal of the project: Investigation of the process of energy accumulation and its transformation to the energy of accelerated particles during solar flares, study of the acceleration mechanisms, propagation and interaction of the fast particles in the solar atmosphere.

Study of the solar activity correlation with physics-chemical processes in Earth upper atmosphere

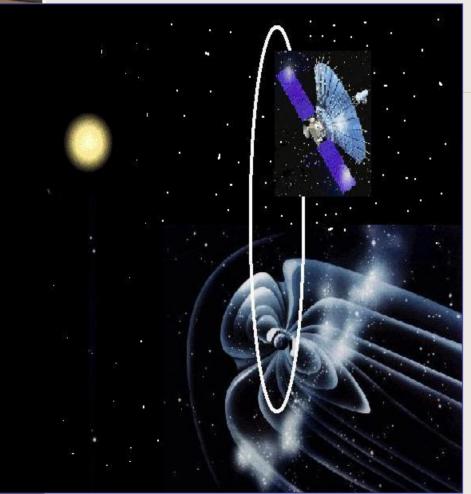
"CORONAS-PHOTON"

| MAIN CHARACTERISTICS OF THE SPACECRAFT | |
|---|-----------------|
| Spacecraft weight, kg | 1900 |
| Scientific payload weight, kg | 540 |
| Orbit: | |
| • type | circular |
| • height, km | 500 |
| • inclination, deg | 82.5 |
| Accuracy of the spacecraft axis orientation to the Sun, arc min | better than 5 |
| Determination accuracy of axis orientation on the Sun, arc min | 3 |
| Angular velocity stabilization of the spacecraft, deg/s | less than 0.005 |
| Volume of scientific information stored per day, Gbit | 8.2 |
| Information transmitted during one communication session, Mbit | t 2048 |
| Nominal mission lifetime, years | at least 3 |
| Information transmitted during one communication session, Mbit | t 2048 |

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http://www.astro.mephi.ru/english/e_photon.htm

Solar-terrestrial payload onboard SPECTR-R



PLASMA-F with participation of China, Czechia, Greece, Slovakia, Ukraine.

SPECTR-R is

an International space VLBI project of Russian Space Agency.

A 10-meter radio telescope will be launched in late **2007** to an orbit with

| apogee | 350 000 km, |
|-------------|-------------|
| perigee | 5 000 km |
| inclination | 54°. |

PLASMA-F is solar-terrestiral payload of opportunity onboard SPECTR-R.

<u>Fine-scale solar wind, IMF, SCL structures</u>: Synchronized 32 Hz measurements of magnetic field, solar wind ion density,

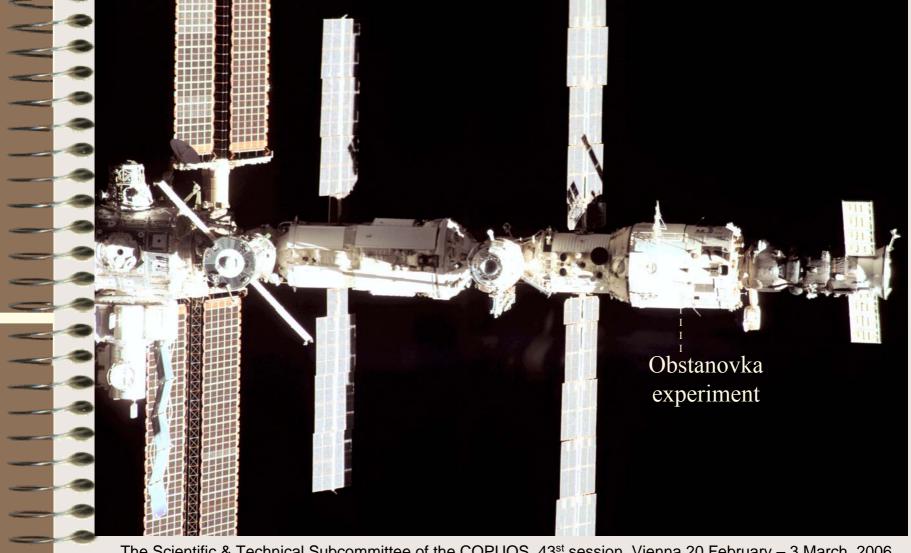
velocity vector, temperature, energetic ions and electrons.

<u>Solar wind, IMF, SCL monitoring</u>: Multi-point solar wind observations with other projects in the frame of ILWS.

www.iki.rssi.ru/obstanovka Experiment « Obstanovka 1-st stage » on the Russian Segment of the ISS.

DFKG









PWC composition

Unit

Combined wave sensor – CWS-1, CWS-2 Flux gate magnetometer – DFM-1 Flux gate magnetometer – DFM-2 Langmuir probe - LP-1, LP-2 Spacecraft potential monitor - DP-1, DP-2 Plasma discharge stimulator – SPP Correlating Electron Spectrograph (10eV – 10KeV) CORES Radio Frequency Analyzer – RFA (Scorpion)

Signal Analyzer and Sampler – SAS3 Data Acquisition and Control Unit - DACU-1, DACU-1 Block of Storage of Telemetry Information – BSTM Grounding support equipment – GSE Booms PWC integration

Responsible Institute LC ISR, Ukraine IKI RAS, Russia LC ISR, Ukraine STIL, **Bulgaria** IKI BAS, **Bulgaria** IKI RAS, Russia Sussex University, UK SRC, Poland SISP, <u>Sweden</u>; SRG, BLE, Hungary KFKI RMKI, Hungary; KFKI RMKI, Hungary; KFKI RMKI, Hungary; RSC "Energia", Russia IKI RAS, Russia



The purposes of the "Obstanovka 1st stage" experiment are:

- geophysical studies of the plasma- wave processes, connected with the manifestation in the ionosphere it is solar - magnetospherically - ionospherically - it is atmospheric - terrestrial connections;
- the ecological monitoring of the low-frequency electromagnetic radiations of anthropogenic nature and connected with the global catastrophes;

• the coordinated ground observations on the influence of electromagnetic disturbances on the technogenic structures and the living organisms.



Coordinated Ground Observations Program

The development of the Program accompanying the "Obstanovka" experiment started in 2005

Regional planning meeting for the Balkan and Black Sea region (<u>http://www.stil.bas.bg/IHY/</u>) recommended to organize the coordinated ground observations

Bulgaria, Armenia, Azerbaijan, Croatia, Georgia, Greece, Poland, Romania, Russia, Serbia and Montenegro and Ukraine



Program of the Space Education, realized on RS ISS, was presented on the EOEM Topical Group Meeting (ESA). For education purpose the part of the physical parameters, measured onboard spacecrafts, will be transferred directly to the School.

The results of space studies are used in many applications, including the education.

Work with the schools is a natural method to inform the general public about the role and value of space studies for humanity.

nttp://www.energia.ru/english/energia/sci-education/microsat/microsat-02.html



The first Russian-Australian scientific-educational microsatellite "Kolibri -2000" (total mass of 20,5 kg), on 20 March, 2002, has been injected Into orbit of International Space Station (ISS) by separation from the transport vehicle "Progress".

nttp://www.energia.ru/english/energia/sci-education/microsat/microsat-02.html





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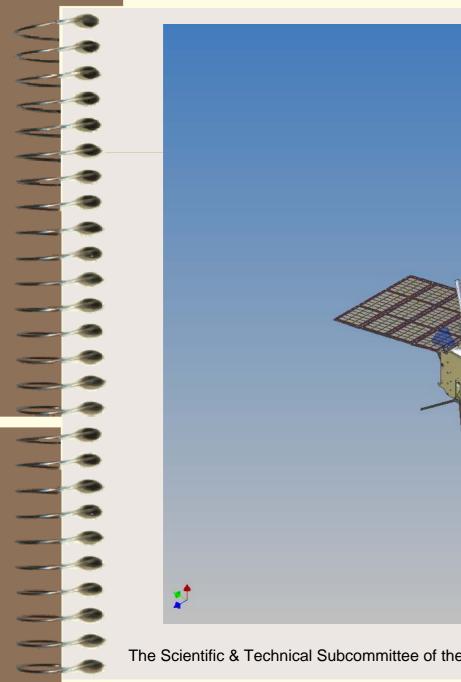
It began the development of tasks for scientific-educational microsatellite (SEM). In spite of small size, SEM had 3.6 kg of scientific payload, which provides an opportunity to carry out rather wide scientific studies both in the field of "classical" space physics and for the space weather, atmosphereionosphere connections etc., it serves also for the tasks of space education. The Scientific & Technical Subcommittee of the COPUOS, 43st session, Vienna 20 February - 3 March 2006

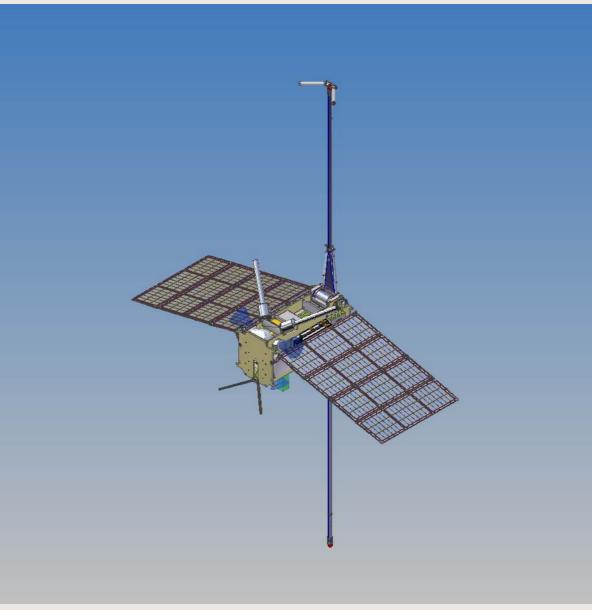
nttp://www.energia.ru/english/energia/sci-education/microsat/microsat-02.html



In IKI RAN is finished the phase "A" – is developed the model composition of the complex of scientific instruments, support systems, construction of microsatellite "Chibis".

Microsatellite is executed with the use of an experience of microsatellite "Kolibri-2000"





КИ

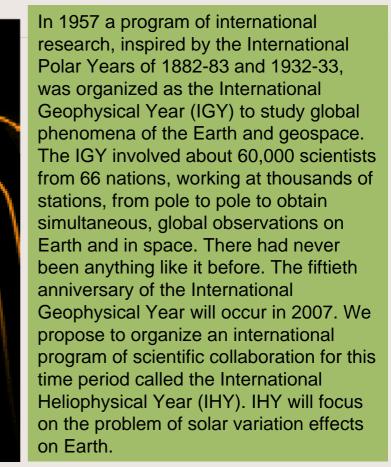


•Main technical characteristics of the micro-satellite ''Chibis''.

| • <u>Main technical characteristics of the micro-satelli</u> | te "Chibis". |
|--|---------------------------------------|
| | |
| • Mass | - 40 kg. |
| Scientific instruments | - 12.5 kg. |
| Service system | - 18.2 kg. |
| Construction and temperature control system | - 9.3 kg. |
| - circular w | ith the height ~ of 480 km. |
| | <u> </u> |
| •Orientation systems: | |
| - types: the electromechanical (electroflywheel | s) magnetodynamic |
| electromagnets) gravitational (boom); | , 6 , |
| - accuracy of the determination of orientation f | rom the sensors (starry, |
| solar) and systems GPS - GLONASS | to 2- angl. deg. |
| - accuracy of guidance | +/- 3 - 15 angl. deg. |
| Data tuangmission systems | |
| •Data-transmission system: - S/C-Earth | - 128 kbit/s |
| | |
| - the capacity of onboard storage | - 8 Mbytes |
| - the volume of the adopted from the board information of the sector of command and telemetering | · · · · · · · · · · · · · · · · · · · |
| The radio frequency of command and telemeterin | 8 |
| The system of onboard power supply | 50 W: |

INTERNATIONAL HELIOPHYSICAL YEAR 2007=1957 (IGY)+ 50

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IHY+**IPY** (INTERNATIONAL POLAR YEAR)



Thanks for the attention Distinguished Chairman and the respected Delegates