#### Lares System: a succesfull example of low cost high science mission

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Wien 18 February 2013

VV 01 • LARES - CubesSats - ALMASat-1 - February, 13th 2012



The Heritage

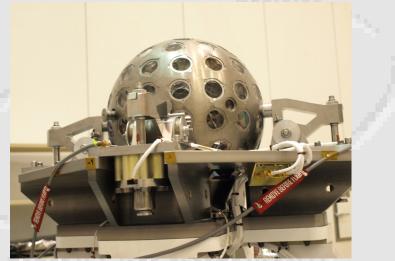


#### In 1992 AS launched with the IRIS launch system on board STS 52 the LAGEOS-2.



It has a diameter of 60 cm, mass 405 kg and orbits earth at 5900 km. Those are perfect parameters for Geodesy and also provide some indications on relativistic effects, but is non what is needed for reach the necessary accuracy for an exact determination of the Lense – Thirring effect. OPProm LAGEOS2 to LARES

- The LAGEOS limitations for fundamental physics had lead already in mid 90' the team guided by Prof |. Ciufolini and Prof A. Paolozzi to developed a specific design for a Laser Relativity Satellite: LARES
- The satellite should have some specific characteristics : it was to be small and very heavy in order to reduce the so called non-gravitational forces
- LARES principal Parameters
- Mass 386,8 Kg
- Diameter 36 cm
- Mirrors 92 CCR
- Orbit 1450 Km, 69.5°
- Goal: measure Lense-Thirring effect with an accuracy of 1%





- In 2008 ASI agreed with ESA to embark on board the VEGA maiden flight a scientific payload
- This opportunity was recognized should need a design to cost approach strictly tailored with the scope of the scientific mission and the risk of a maiden flight
- The original contract foresaw a launch in 2009
- The following elongation of the Vega program allowed to modify the design in order to host on the LARES System other payloads: 9 CubeSats + Almasat

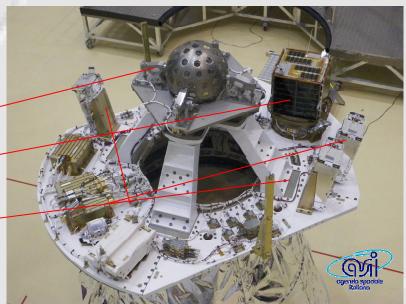
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Lares Platform Lares satellite Almasat Cubesats



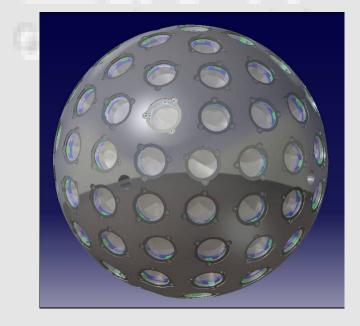
The Lares System provides also additional Vega environmental data and images of the flight and satellites separations









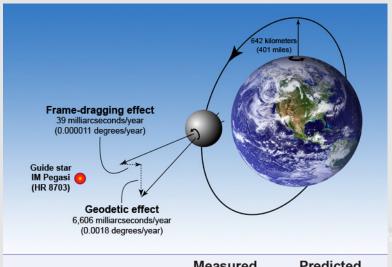






### Manual Science: The frame dragging

- Any mass induce a deformation in the Space-Time frame
- If the mass rotates then also the ST frame co-rotates and a mass in this frame will be dragged
- This effect, predicated by Einstein theory is called Lense – Thirring effect
- The accurate measure will provides benefit to the studies on relativity and also on current applications as GPS



|                              | Weasureu   | Fledicied | _ |
|------------------------------|------------|-----------|---|
| Geodetic precession<br>(mas) | 6602 ± 18  | 6606      |   |
| Frame-dragging<br>(mas)      | 37.2 ± 7.2 | 39.2      |   |

Gravity Probe results

# OPALARES: a satellite of records

- 1. The orbiting object with the highest mean density in the solar system
- 2. The heaviest tungsten alloy piece ever manufactured
- The orbiting artificial object with the lowest surface-tomass ratio
- 4. The satellite body with the lowest number of parts: 1
- 5. satellite built in 4 year and 1 week
- 6. Among the longest incubation period 1984-2008 (24 years)

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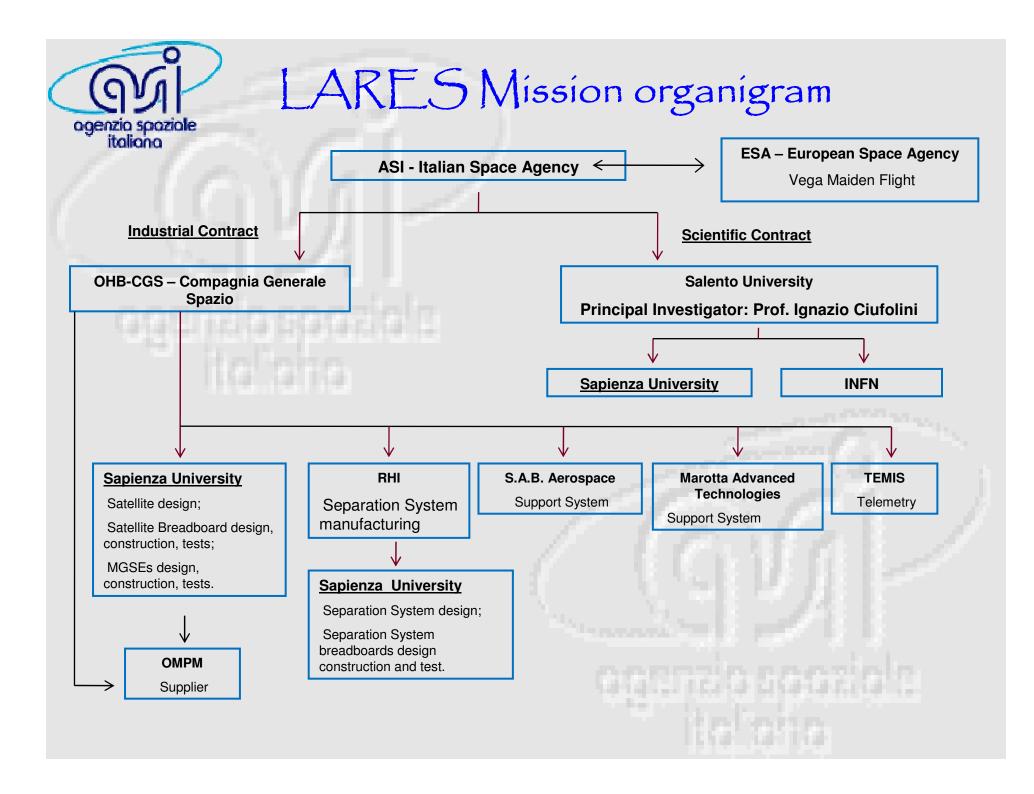
Design to cost approach

Low cost : less then 6.5 M€ = 1/5 of the launcher cost

- Extensive use of COTS elements
- Spin-in from High-Tec commercial small industry

1720 20020

 Strong cooperation between Industry and University





C1-To deploy: a) Lares satellite b) Almasat c) Cubesats C2-To acquire : a) images of the VEGA and satellites separations b) Vega Flight environmental data



C2 - ] a) VE

b) Al

Lares System Results



LARES Classical Orbit Elements Time (UTCG): 14 Feb 2012 16:49:15.828 Semi-major Axis (km): 7818.315676 Eccentricity: 0.000344 Inclination (deg): 69.429 RAAN (deg): 236.483 Arg of Perigee (deg): 263.623 True Anomaly (deg): 149.111 Mean Anomaly (deg): 149.091

| AVUM Classical Orbit Elen | ments    |       |         |
|---------------------------|----------|-------|---------|
| Time (UTCG): 14           | Feb 2012 | 16:49 | :15.828 |
| Semi-major Axis (km):     | •        | 7233  | .161548 |
| Eccentricity:             |          | 0     | .080862 |
| Inclination (deg):        |          |       | 69,458  |
| RAAN (deg):               |          |       | 235.881 |
| Arg of Perigee (deg):     |          |       | 45.822  |
| True Anomaly (deg):       |          |       | 343.507 |
| Mean Anomaly (deg):       |          |       | 345,992 |
|                           |          |       |         |

0 35 02 27 ssociato al lancio 2012o "Analyst", mentre alla

ES in risposta all'ODR

a ellittica finale di

Earth Inertial Axes 14 Feb 2012 16:49:15.82

15.828 Real Time Offset: 0.00 sec



#### Signal correctly acquired four days after launch





Lares Program

## All Criteria Met!

# Science Phase











