







# **Space Navigation in Russia: History of Development**

# **TESTOYEDOV Nikolay**

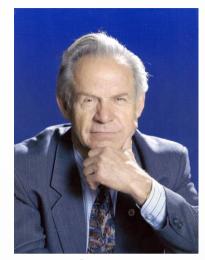
**General Director** 

JSC "Academician M.F.Reshetney "INFORMATION SATELLITE SYSTEMS"



# START OF SPACE NAVIGATION IN RUSSIA





RESHETNEV Mikhail

Hero of Socialist Labor, Laureate of the Lenin Prize and the State Prize, Academician of USSR Academy of Science (Russian Academy of Sciences) Development of space activity in Siberia started on June 4, 1959 with the affiliated company of Design Bureau No.1 established in the city of Krasnoyarsk–26 (today, city of Zheleznogorsk).

The first practical implementation of the company was the 11K65 launch vehicle ("Cosmos") and three small satellites of the "Cosmos" series (No. 38,39,40) launched on 18 August, 1964 which successfully continues today.

1 August 1977 - NPO PM was established.

1997 - NPO PM was renamed as "Academician M.F. Reshetnev NPO PM".

3 March 2008 - rearranged as JSC «Academician M.F. Reshetnev "Information Satellite Systems".









# SATELLITE PRODUCTION CYCLE



- Development
- Manufacture
- Testing
- Maintenance

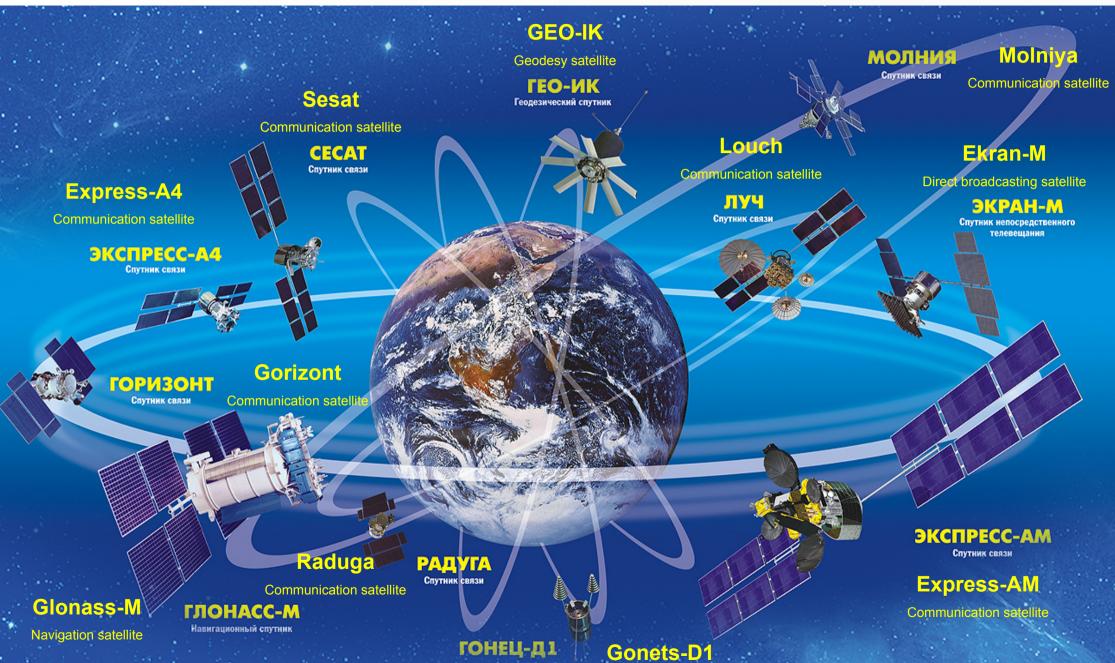
of satellites and systems designed for communication, TV-broadcasting, data relay, navigation, and geodesy for the benefit of national security, social and economic development, and cultural evolution of the country and international links





# SATELLITES FOR ALL ORBIT TYPES

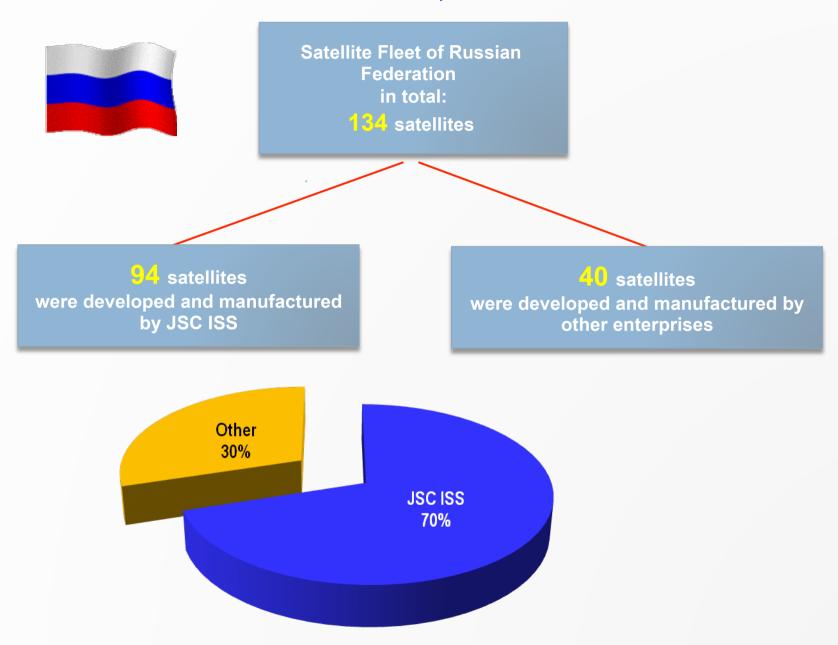






# STATUS OF THE RUSSIAN ORBITAL FLEET **AS OF MAY 18, 2015**









# JSC «ACADEMICIAN M.F. RESHETNEV « INFORMATION SATELLITE SYSTEMS» (ZHELEZNOGORSK) IS THE PIONEER OF SATELLITE **NAVIGATION IN RUSSIA**



# **SPACE NAVIGATION**







1967

# **NAVIGATION SATELLITE EVOLUTION**









Glonass GLONASS-M

2nd Generation of SNS





3d Generation of SNS



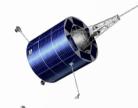


#### FIRST GENERATION OF THE SPACE NAVIGATION SYSTEM



Satellite navigation started with creation of space navigation systems based on Doppler navigation approach. With a Doppler frequency increment for one fast-moving satellite, an user defines his own position on the ground surface (which is especially for benefit of maritime users) with intervals of ~ 1.5 hours and accuracy of 100 meters.

These parameters determined the orbit type selected (circular subpolar orbit with altitude of ~ 1000 km) and number of satellites (~ 6).



#### SNS based Tsiklon S/C

The first launch of Tsiklon No. 11F (Cosmos-192) – November 23,1967 Total **25** Tsiklons

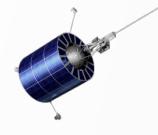


#### SNS based Tsiklon-B S/C (the Parus (Sail) System)

The first launch of Tsiklon-B No.11F (Cosmos -700) — December 26, 1974 Total 93 Tsiklons-B

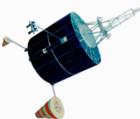


Total 21 Tsikadas



#### SNS based Nadezhda S/C (COSPAS-SARSAT):

The first launch of Nadezhda No.11F (Cosmos-1383) – June 30, 1982 Total **10** Nadezhdas

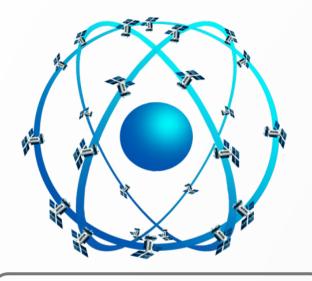




# **GLONASS SYSTEM SPACE COMPLEX**



#### **GLONASS SPACE COMPLEX ARCHITECTURE**





Generation and transmission of navigation signals

Orbit: circular, H=19140 km, I=64.8 degrees





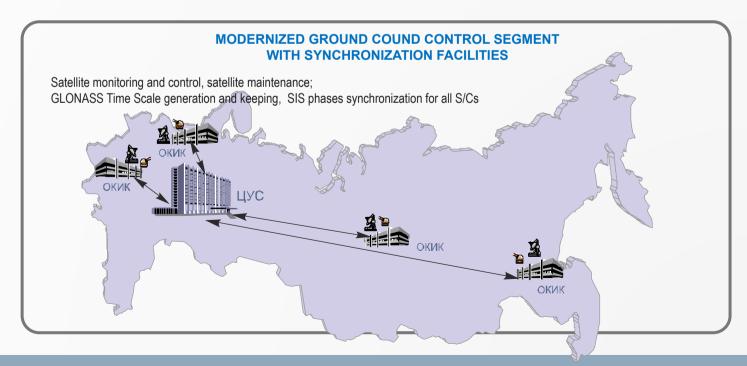


#### **LAUNCH VEHICLES**

S/C launches, development and replenishment of the nominal orbital constellation







**GLONASS CONSTELLATION** 



# **SECOND GENERATION OF THE SPACE NAVIGATION SYSTEM**







| Design life                               | - 3 years;              |
|---|-------------------------|
| Mass                                      | - 1413 kg;              |
| Power consumption                         | - 1000 W;               |
| Accuracy of S/C Earth orientation         | - ± 0,5 deg.;           |
| Accuracy of S/C Sun orientation           | - ± 2 deg;              |
| Stability of onboard frequency generator  | - 1×10 <sup>-13</sup> ; |
| Navigation signals:                       |                         |
| Frequency Division Multiple Access - L10F | : L1SF: L2OF: L2SF:     |

- 88 S/Cs

| Design life                          | - 7 years;                   |
|--------------------------------------|------------------------------|
| Mass                                 | - 1415 kg;                   |
| Power consumption                    | - 1250 W;                    |
| Accuracy of S/C Earth orientation    | - ± 0,5 deg;                 |
| Accuracy of S/C Sun orientation      | - ± 2 deg.;                  |
| Stability of onboard frequency gener | ator - 1×10 <sup>-13</sup> ; |
| Navigation signals:                  |                              |
| Frequency Division Multiple Access   | - L10F; L1SF; L2OF; L2SF;    |
| Launched                             | - 42 S/Cs                    |

Launched



# THIRD GENERATION OF THE SPACE NAVIGATION SYSTEM





| Design life                              |
|--|
| Mass                                     |
| Power consumption                        |
| Accuracy of S/C attitude control         |
| Stability of onboard frequency generator |
| Navigation signals:                      |
| - FDMA                                   |
|  |

| ,            |             |            |         |  |
|--------------|-------------|------------|---------|--|
| Stability of | onboard fre | equency ge | nerator |  |
| Navigation   | signals:    |            |         |  |
| - FDMA       |             |            |         |  |
|              |             |            |         |  |
| - CDMA       |             |            |         |  |
| Launched     |             |            |         |  |
|              |             |            |         |  |
|              |             |            |         |  |

| - 10 years;             |
|-------------------------|
| - 962 kg;               |
| - 1600 W;               |
| - ± 0,5 deg.;           |
| - 5×10 <sup>-14</sup> ; |
|                         |
| - L10F; L1SF;           |
| L2OF; L2SF;             |
| - L3OC;                 |
| - 2 S/Ca                |
|                         |

Launch



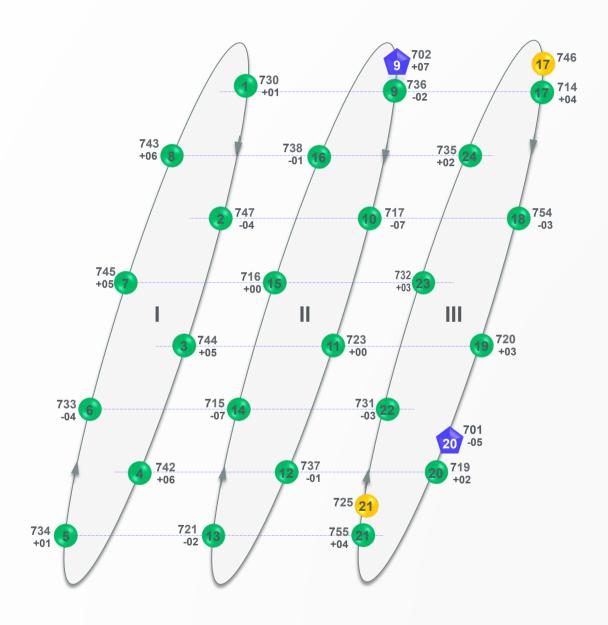
| Design life                              | - 10 years;                                 |
|--|---|
| Mass                                     | - 1645 kg;                                  |
| Power consumption                        | - 4370 W;                                   |
| Accuracy of S/C attitude control         | - ± 0,25 deg;                               |
| Stability of onboard frequency generator | - 5×10 <sup>-14</sup> ÷ 5×10 <sup>-15</sup> |
| Additional PL equipment                  | - 6   |
| Navigation signals:                      |   |
| - FDMA                                   | - L10F; L1SF;                               |
|  | L2OF; L2SF;                                 |
| - CDMA                                   | - L10C; L1SC;                               |
|  | L2SC; L3OC;                                 |
|  |   |

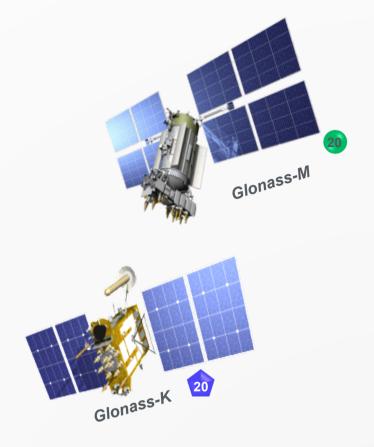
- since 2018



# **GLONASS CONSTELLATION**





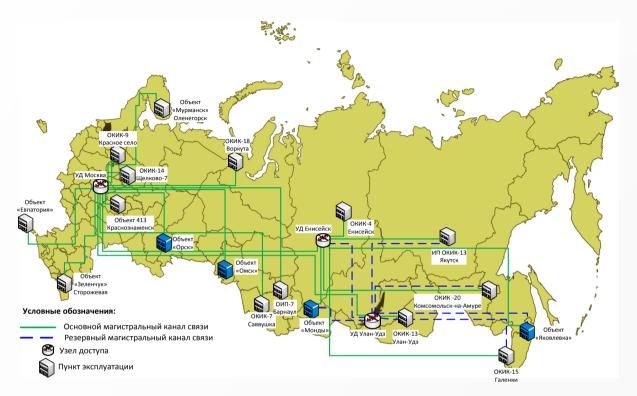


| Status as of May 18, 2015. |         |  |  |  |  |  |  |  |
|----------------------------|---------|--|--|--|--|--|--|--|
| Total                      | 28 S/Cs |  |  |  |  |  |  |  |
| Operational                | 24 S/Cs |  |  |  |  |  |  |  |
| Under flight test          | 2 S/Cs  |  |  |  |  |  |  |  |
| Under investigation        | 2 S/Cs  |  |  |  |  |  |  |  |



### **GROUND CONTROL SEGMENT**





The Ground Control Segment (GCS) serves for automated control of GLONASS satellites

Within the management process GCS resolves the following tasks:

- Planning of work with satellites for GCS means, control of satellite injected into orbit;
- Control and analysis of satellite operation and status;
- Determination and prediction of satellite orbit parameters and calculation of ephemeris, clock, and almanac data;
- Upload and control of mission data (ephemeris, clock data, etc.) with the purpose of navigation frame generation;
- Check, phasing, and correction of Onboard Time Scale;
- Control of navigation data downloaded;
- Control of SIS performances.



# PROGRAM OF THE GLONASS CONSTELLATION DEVELOPMENT AND SUSTAINMENT

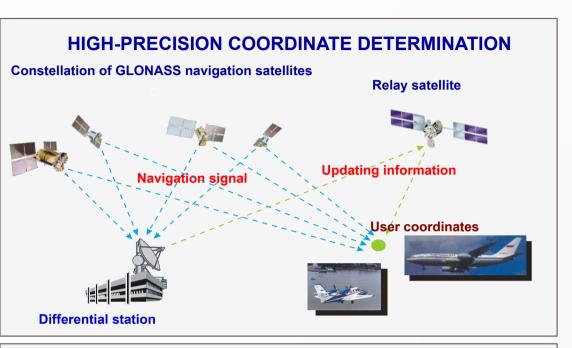


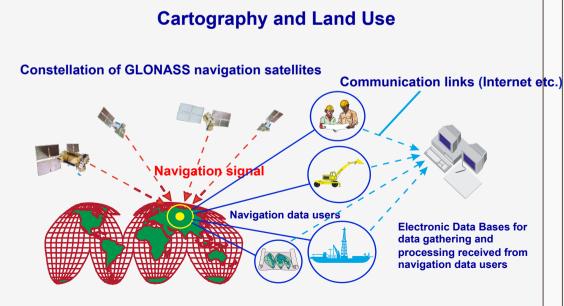
|     | 20  | 14  |     |    | 20  | 15              |     |     | 20 | 16  |     | 2017 |    |     | 2018              |            |     |   | 2019              |                   |    |      | 2020 |      |    |     |      |
|-----|-----|-----|-----|----|-----|-----------------|-----|-----|----|-----|-----|------|----|-----|-------------------|------------|-----|---|-------------------|-------------------|----|------|------|------|----|-----|------|
| I   | II  | III | IV  | I  | II  | III             | IV  | I   | II | III | IV  | I    | II | III | IV                | I          | II  | III   | IV                | ı                 | II | III  | IV   | ı    | II | III | IV   |
| 24  | 24  | 24  | 24  | 24 | 24  | 24              | 24  | 24  | 24 | 24  | 24  | 24   | 24 | 24  | 24                | 24         | 24  | 24  | 24                | 24                | 24 | 24   | 24   | 24   | 24 | 24  | 24   |
| 54  | 55  |     | K12 |    | 51  | 52<br>53<br>56  | 57  | 58  |    | 59  | 60  |      |    | 61  | К <sub>2</sub> 13 | K13<br>K14 | K15 | K16<br>K17<br>K18   | K <sub>2</sub> 14 | K19<br>K20<br>K21 |    | К22  |      | К23  |    |     | K₂15 |
|     |     |     |     |    |     | Teanor of Teans |     |     |    |     |     |      |    |     |                   |            |     | Teachor of |                   | TEAO-OR O         |    |      |      |      |    |     |      |
| 48c | 49c |     | К2с |    | 50c | 51              | 52c | 53c |    | 54c | 55c |      |    | 56c | КЗс               | K4c<br>K5c | К6с | К7  | К8с               | К9                |    | К10с |      | К11с |    |     | K12c |

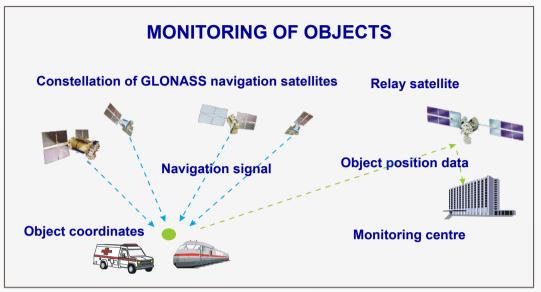


# HIGH PRECISION INFORMATION PROVISION









#### **Monitoring of Extended Objects Condition**



The project of the motorway bridge structural monitoring for M-53 federal highway near Krasnoyarsk was designed and is in use currently.

Technology of bridge member control using GLONASS technology was proven under real-time testing.

Accuracy characteristics for relative motion of bridge reference points: 1-3 mm



# SPACE SYSTEM FOR AUTOMATED OIL PIPELINE MONITORING USING GLONASS GNSS



(one of the prospective projects)

