

# QB50

## An international network of CubeSats

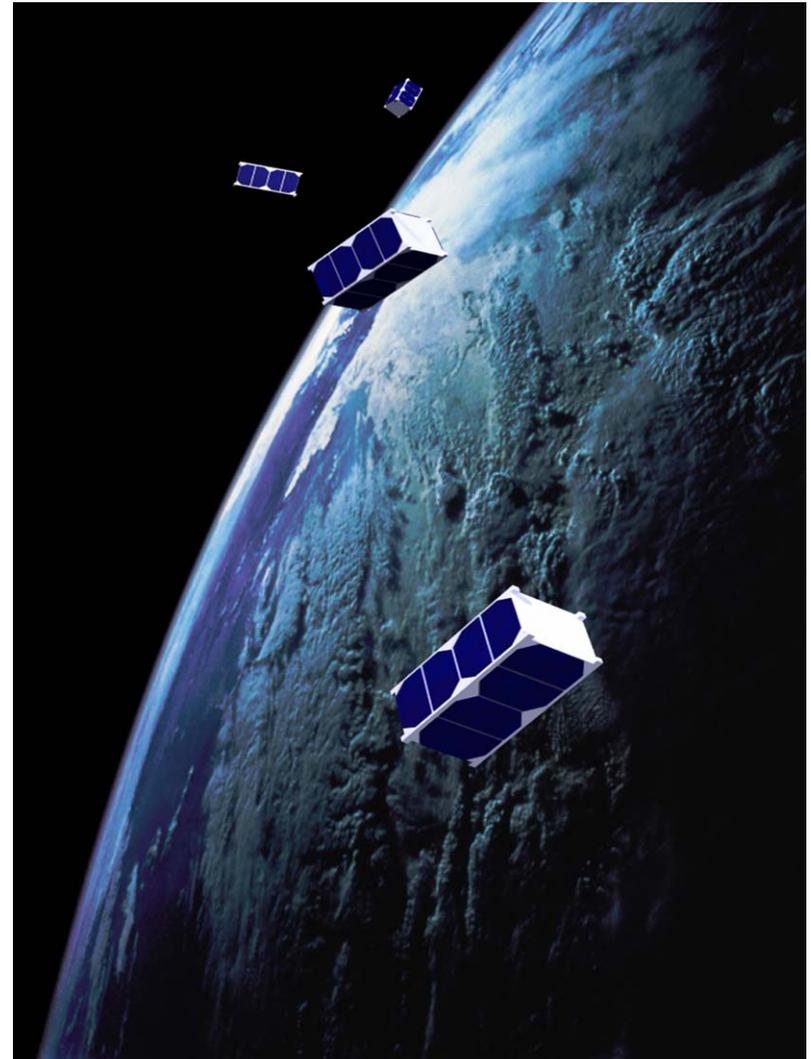
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von Karman Institute for Fluid Dynamics  
Rhode-Saint-Genèse (Brussels)

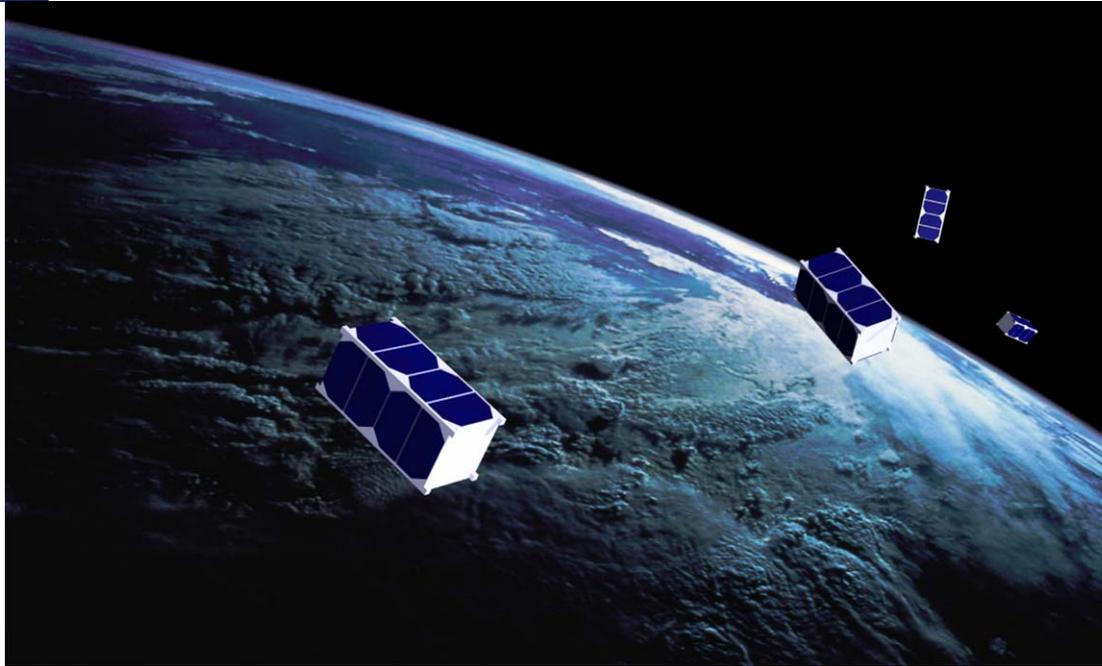
**UNCOPUOS**

**Technical and Scientific Committee**

15 Feb 2013  
Vienna , Austria



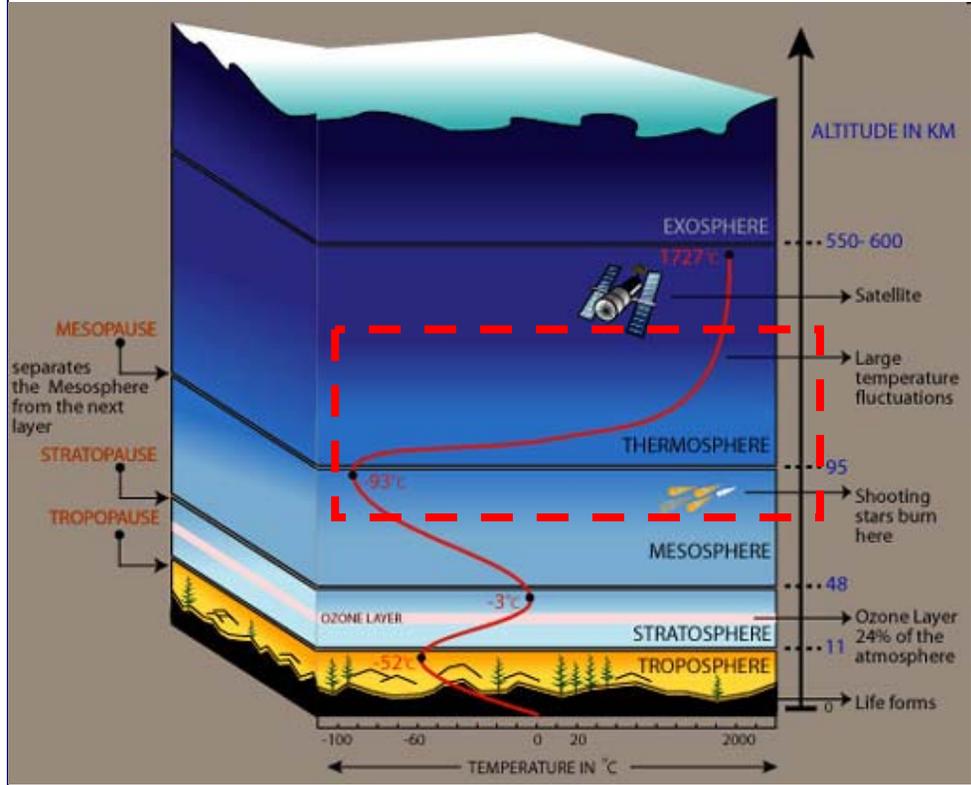
# QB50 - THE IDEA



- An international network of 50 CubeSats for multi-point, in-situ, long-duration measurements and in-orbit demonstration in the lower thermosphere
- A network of 50 CubeSats sequentially deployed
- Initial altitude: 350 km (circular orbit, high inclination)
- Downlink using the QB50 Network of Ground Stations

## 90 – 330 km: Why Lower Thermosphere?

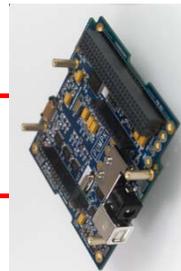
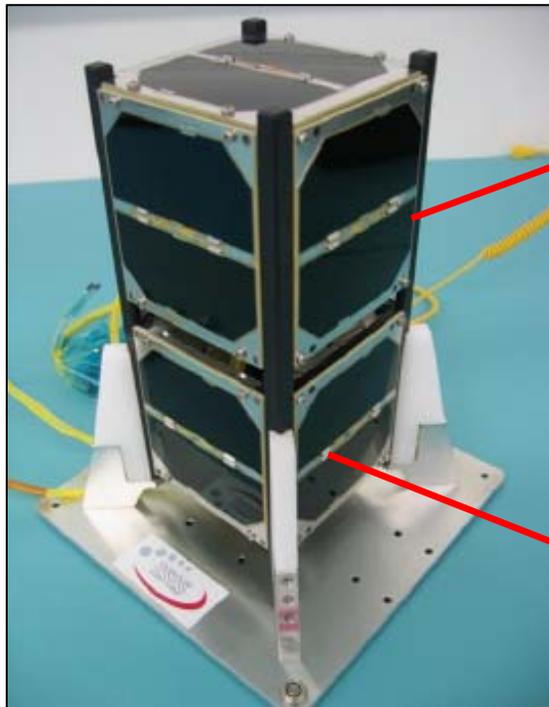
- The **least explored** layer of the atmosphere
- Stratospheric balloons go up to 42 km max.
- Remote-sensing by ground based lidars and radars up to 105 km.
- Remote-sensing by Earth observation satellites in higher orbits (600 – 800 km) only observe constituents in the troposphere, stratosphere and mesosphere (lower thermosphere is too rarefied).
- In-situ measurements by sounding rockets in the mesosphere and lower thermosphere (MLT Region) provide only occasional (a few times per year) single-line measurements



# QB50 - The CubeSat



***On a Double CubeSat (10 x 10 x 20 cm<sup>3</sup>):***

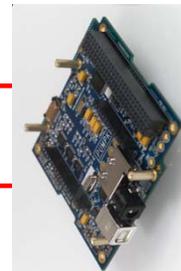


## **Science Unit:**

*Lower Thermosphere Measurements*

*Sensors designed by MSSL*

*Standard sensors for all CubeSats*



## **Functional Unit:**

*Power, CPU, Telecommunication*

*Optional Technology or Science Package*

*Universities are free to design the functional unit*

# Sensor Selection

## Set 1

- Ion-Neutral Mass Spectrometer (INMS)
- 2 corner cube laser retroreflectors (CCR)\*
- Thermistors/thermocouples/RTD (TH)

FIPEX sensor



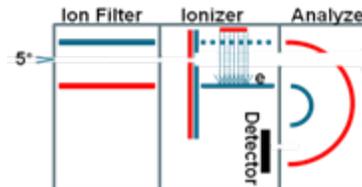
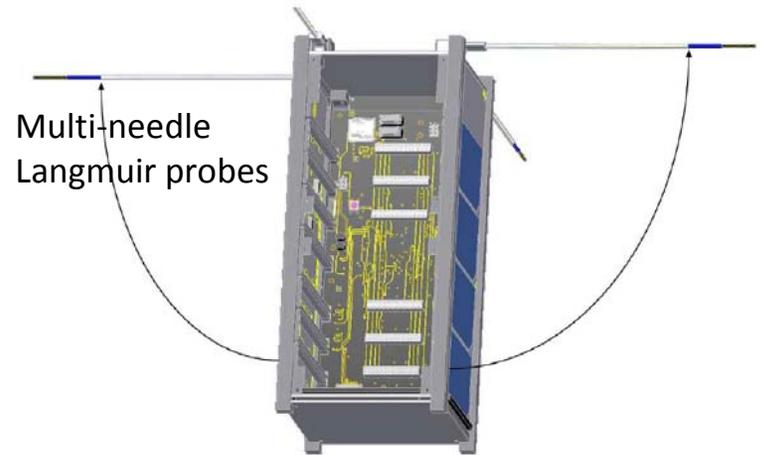
## Set 2

- Flux- $\Phi$ -Probe Experiment (FIPEX)
- 2 corner cube laser retroreflectors (CCR)\*
- Thermistors/thermocouples/RTD (TH)

## Set 3

- A set of 4 Langmuir probes (MNLN)
- 2 corner cube laser retroreflectors (CCR)\*
- Thermistors/thermocouples/RTD (TH)

\* Offered as an option

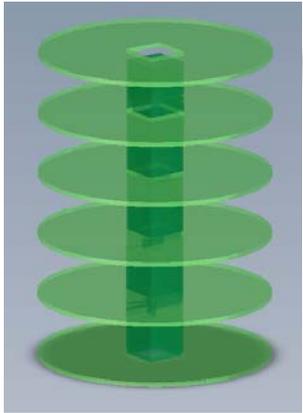


Schematic of the principle of working of the INMS

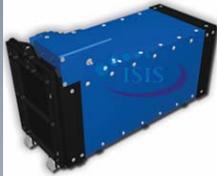


Miniaturised charged particle analyser along with the Improved Plasma Analyser

# In-Orbit Demonstration



A modular deployment system for double and triple CubeSats

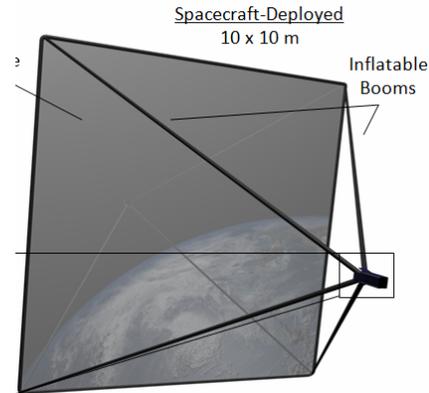
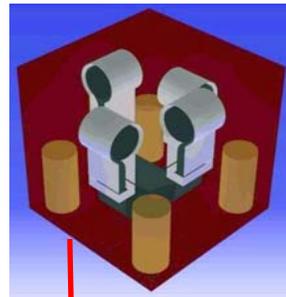


VKI's Re-Entry CubeSat  
QARMAN

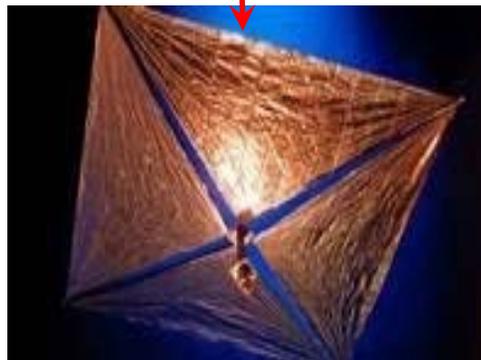
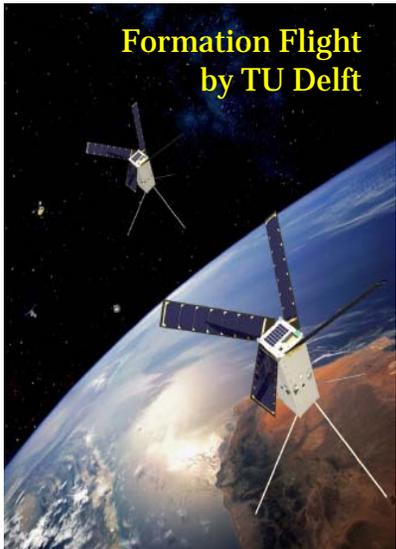
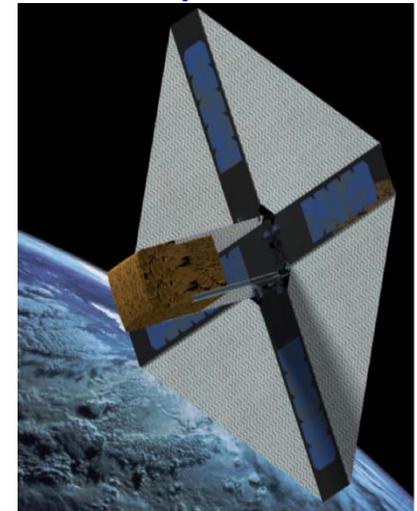
De-orbiting and aerodynamic stability

AeroSDS by VKI

Gossamer-1 Solar Sail demonstration packagem DLR



InflateSail demonstration mission, SSC



## Other In-Orbit Demos:

- End of life analysis, Debris
- Micro-propulsion systems
- Micro-g experiment

# Formation Flying



DeFFI Project: with triple CubeSats “Delta” and “Phi”



- Delft University of Technology intends to provide two triple-unit Cubesats, both being equipped with a highly miniaturized propulsion system in addition to the standard science payload.

- This allows for a coordinated formation flying of these two satellites using baselines, which can be realized, maintained and adjusted during the mission based on scientific and technological needs.

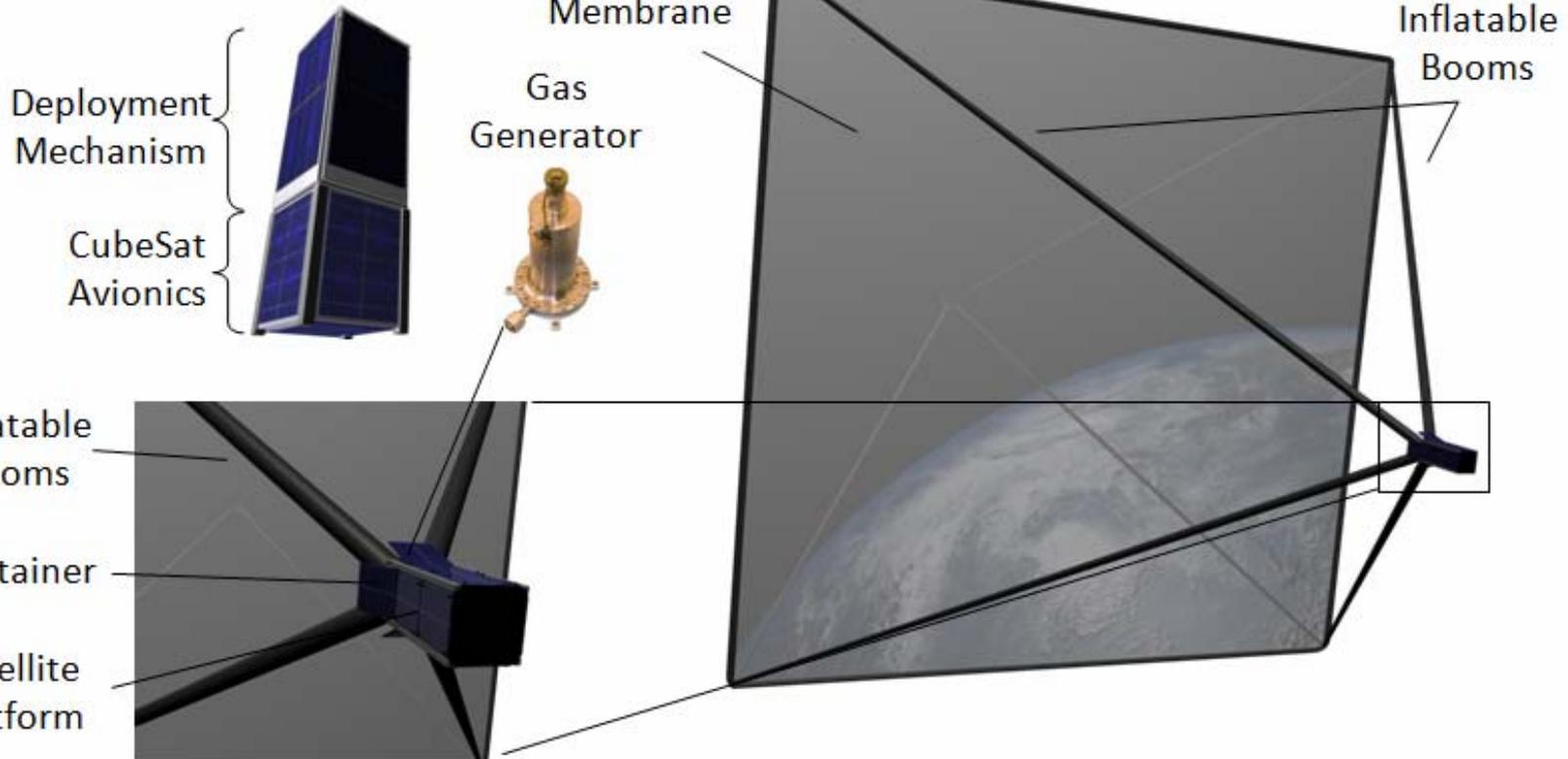
- The position of the satellite will be determined by GPS. The inter-satellite communication will be realized by ground stations
- Therefore, formation flight will be possible at any distance

# Inflate-Sail

for testing a solar sail with inflatable booms

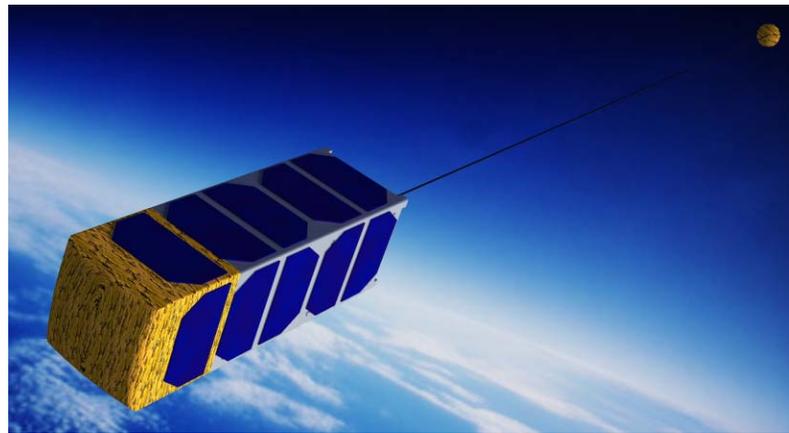
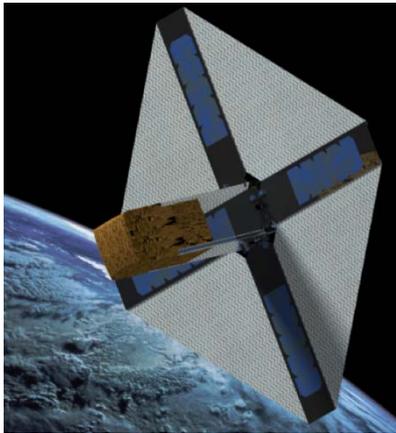
Spacecraft-Stowed  
10 x 10 x 34 cm

Spacecraft-Deployed  
10 x 10 m



# AeroSDS

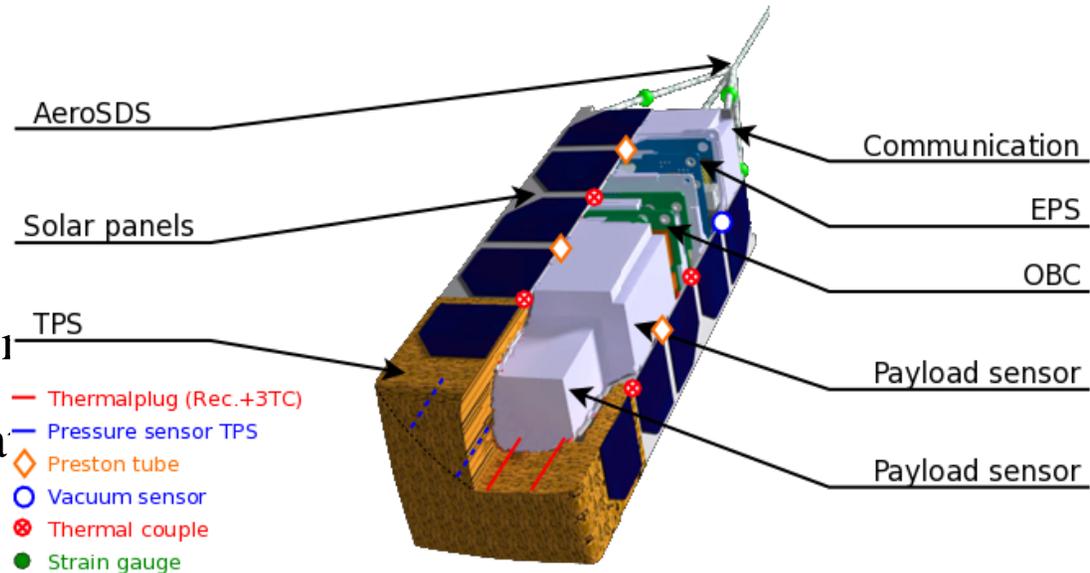
- low-cost, passive and permanent stability
  - powered only during deployment
  - standard COTS systems
- two modes for rarefied & entry phase



- flexible sizing according to desired entry conditions and lifetime

- Measurements for Satellite Re-Entry Trajectory Rebuilding:

- Ablation
- Radiation
- TPS Efficiency
- Shear Force & Transition
- Off-Stagnation Temperature
- Rarified Flow
- Stability

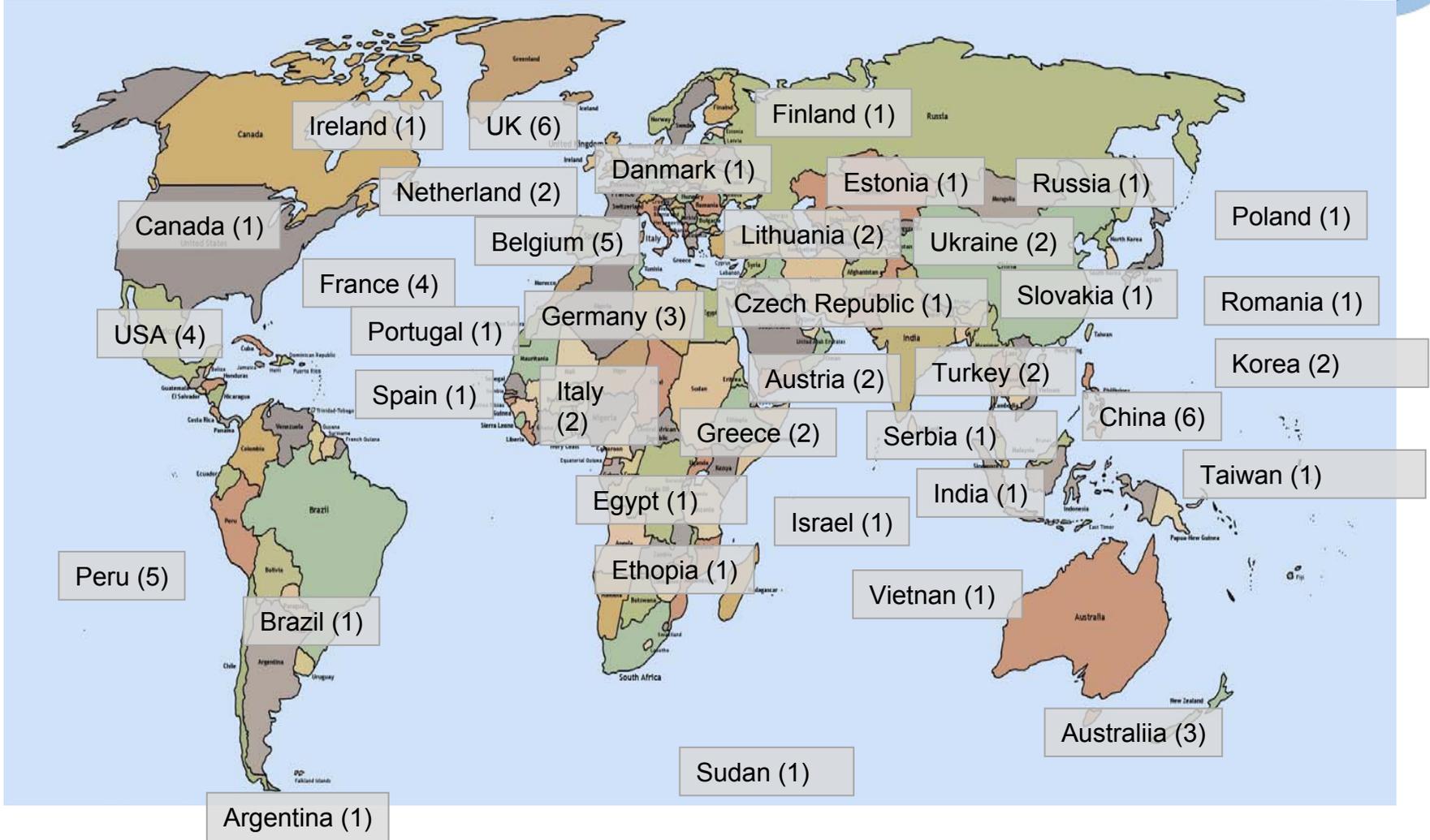


- Feasible Off-the-Shelf Subsystems: IRIDIUM Downlink

- Models for extrapolation of data: **Ground** ⇔ **Flight**

- More than 70 proposals received
- Selection of the 50 CubeSats
  - about 40 double CubeSats for atmospheric research to be selected from 50 proposals,
  - about 10 double and triple In-Orbit-Demonstration CubeSats to be selected from 20 proposals, 4 of them already pre-selected (Delta, Phi, QARMAN, Inflatesail)
- Draft Contractual Agreement between the QB50 Consortium and the proposing universities
- Availability of funding and readiness at the PDR are critical issues in the selection process,
- There will be backup CubeSat teams as well

# QB50 – CU Teams



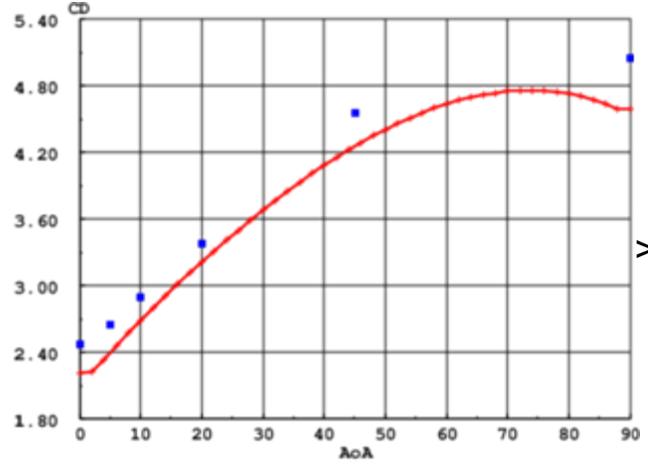
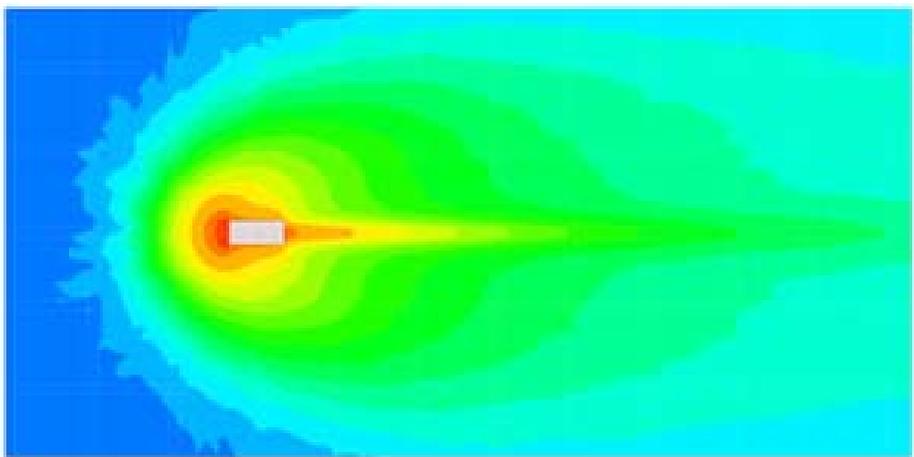
# Status of QB50 Project



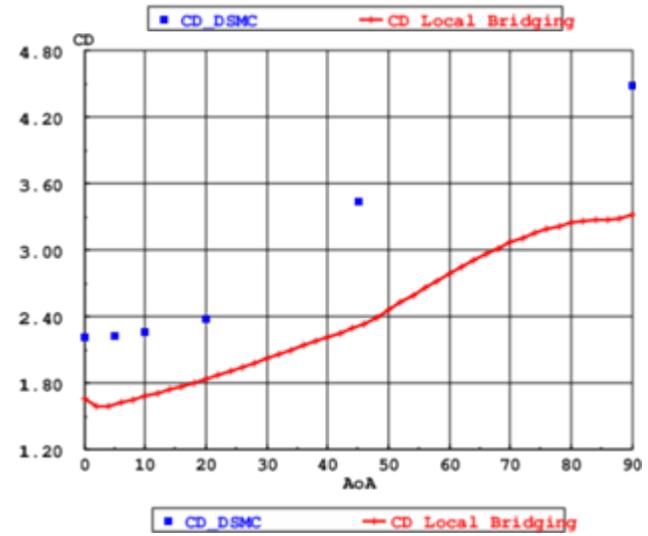
- Started working on the Project as of Nov 2011
- Kick-off was held at 22 Nov 2011
  
- The Call for Proposals issued on the QB50 web site
- More than 70 proposals were received
- Major technical work accomplished on
  - Orbital dynamics
  - Sensitivity analysis on interaction with the atmosphere
  - Deployment strategy
  - Deployment system
  - Science payload design

# DSMC simulations for CubeSat – Atmosphere interaction

Preliminary computations for selected amount of points of re-entry trajectory were performed and aerothermodynamic characteristics of CubeSat were obtained in free-molecular, transitional, and near-continuum flow regimes and accuracy of the engineering methods was assessed by comparison with the results obtained by the DSMC SMILE code (ITAM & VKI)



> 100 km



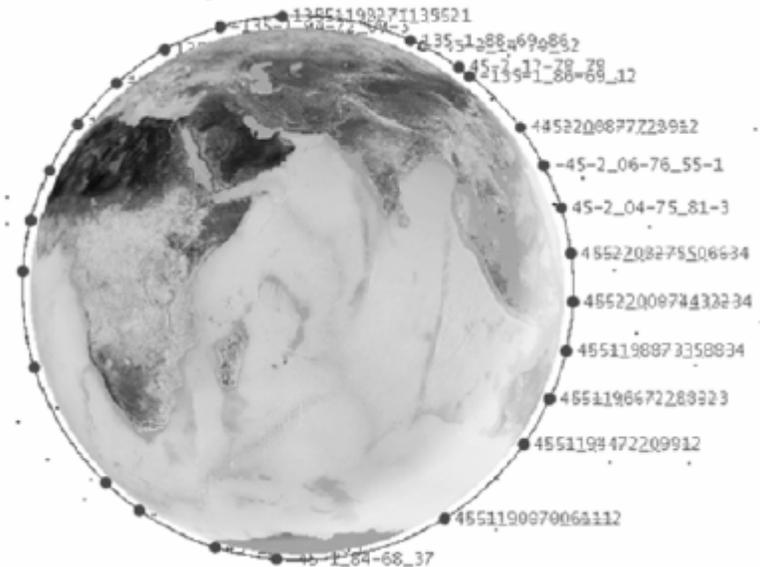
< 80 km

# Deployment Strategy

- How to deploy the 50 CubeSats with minimal collision risk and optimised distribution ?
- Detailed analysis covering ballistic coefficient, deployment direction, deployment frequency
- Best scenario to minimize risk in the first 8 hours, and to optimise a uniform network distribution – the developed strategy can be used directly with the ballistic coefficient database of the selected CubeSats.



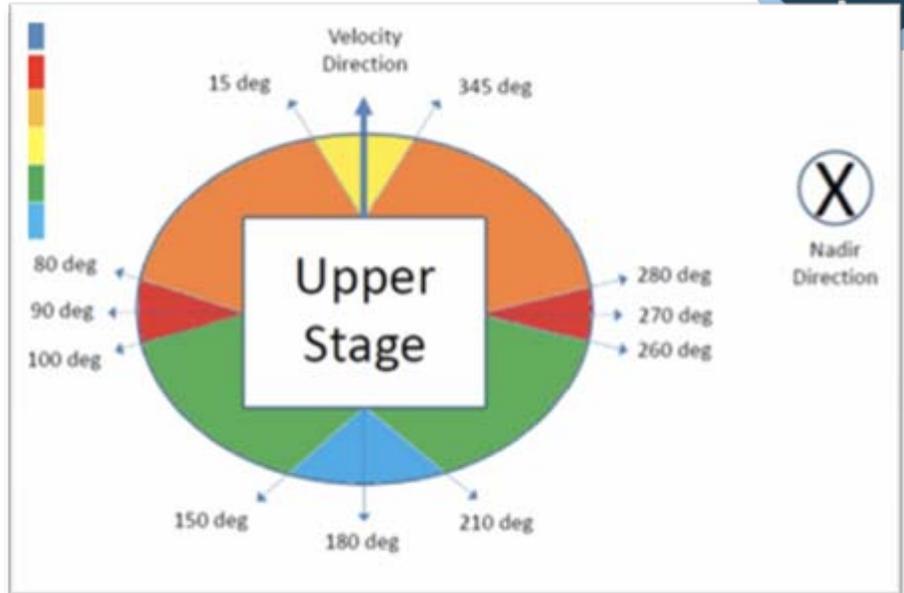
After 20 days



After 30 days

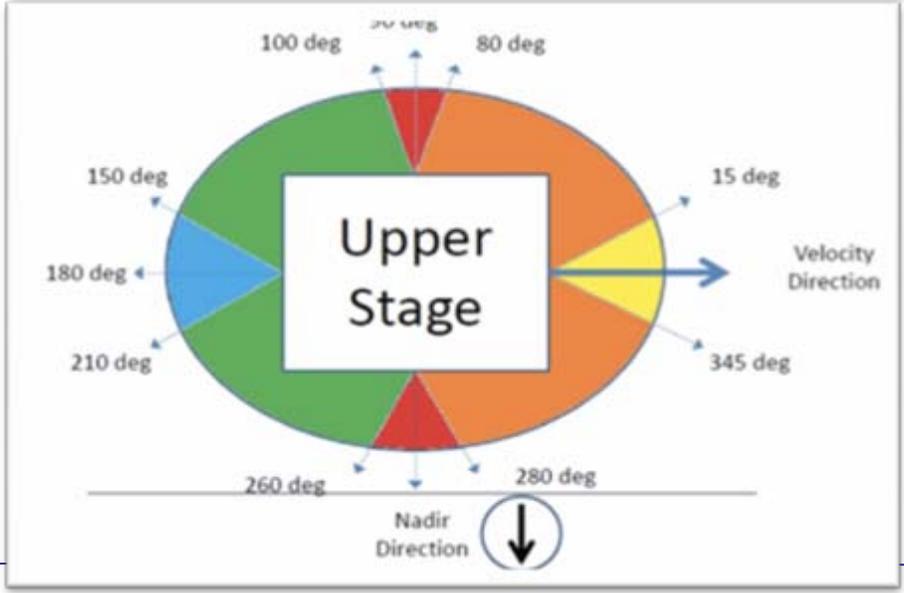
# Deployment Strategy

TopView

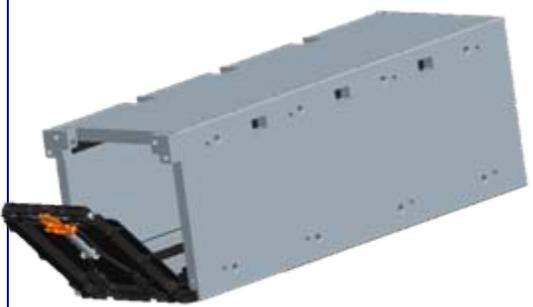


Risk Colors	
High	Red
High-Medium	Orange
Medium	Yellow
Medium-Low	Green
Low	Blue

Side View



# Deployment System



**Concept De-risk  
Prototype  
Prototype**



**Precursor  
Flight  
QuadPack**



**QB50  
StackPack**



# Next Steps



- We are open to further collaboration
  - The 5th QB50 Workshop was held on 29 Jan 2013 at VKI, inviting the CubeSat teams to
    - Signing of the contract with 50 CubeSat providers
    - QB50 requirements documents being processed
    - Get ready for PDR by the end-March 2013
  - Get in touch with our primary Point of Contact  
Cem Ozan Asma, [asma@vki.ac.be](mailto:asma@vki.ac.be)



- [www.CubeSatSymposium.eu](http://www.CubeSatSymposium.eu)
  - Abstract submission: 15 Mar 2013
  - 6 June 2013: 6th QB50 Workshop at VKI, Brussels

# ACKNOWLEDGEMENT

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